

**SYSTEM AND METHOD FOR OPERATING A ROLLOVER CONTROL  
SYSTEM DURING AN ELEVATED CONDITION**

**RELATED APPLICATIONS**

**[0001]** The present invention claims priority to U.S. provisional applications 60/401,416 and 60/401,464 filed August 5, 2002, and is related to U.S. patent applications entitled "SYSTEM AND METHOD FOR DETERMINING AN AMOUNT OF CONTROL FOR OPERATING A ROLLOVER CONTROL SYSTEM" (Attorney Docket No. 202-1221/FGT-1691) and "SYSTEM AND METHOD FOR OPERATING A ROLLOVER CONTROL SYSTEM IN A TRANSITION TO A ROLLOVER CONDITION" (Attorney Docket No. 203-0816/FGT-1869), the disclosures of which are incorporated by reference herein.

**TECHNICAL FIELD**

**[0002]** The present invention relates generally to a control apparatus for controlling a system of an automotive vehicle in response to sensed dynamic behavior, and more specifically, to a method and apparatus for adjusting the activation based on heightened vehicle operating conditions.

**BACKGROUND**

**[0003]** Dynamic control systems for automotive vehicles have recently begun to be offered on various products. Dynamic control systems typically control the yaw of the vehicle by controlling the braking effort at the various wheels of the vehicle. Yaw control systems typically compare the desired direction of the vehicle based upon the steering wheel angle and the direction of

travel. By regulating the amount of braking at each corner of the vehicle, the desired direction of travel may be maintained. Typically, the dynamic control systems do not address rollover (wheels lifting) of the vehicle. For high profile vehicles in particular, it would be desirable to control the rollover characteristic of the vehicle to maintain the vehicle position with respect to the road. That is, it is desirable to maintain contact of each of the four tires of the vehicle on the road.

**[0004]** In vehicle rollover control, it is desired to alter the vehicle attitude such that its motion along the roll direction is prevented from achieving a predetermined limit (rollover limit) with the aid of the actuation from the available active systems such as controllable brake system, steering system and suspension system. Although the vehicle attitude is well defined, direct measurement is usually impossible.

**[0005]** During a potential vehicular rollover event, wheels on one side of the vehicle start lifting, and the roll center of the vehicle shifts to the contact patch of the remaining tires. This shifted roll center increases the roll moment of inertia of the vehicle, and hence reduces the roll acceleration of the vehicle. However, the roll attitude could still increase rapidly. The corresponding roll motion when the vehicle starts side lifting deviates from the roll motion during normal driving conditions.

**[0006]** When the wheels start to lift from the pavement, it is desirable to confirm this condition. This allows the system to make an accurate determination

as to the appropriate correction. If wheels are on the ground, or recontact the ground after a lift condition, this also assists with accurate control.

**[0007]** Some systems use position sensors to measure the relative distance between the vehicle body and the vehicle suspension. One drawback to such systems is that the distance from the body to the road must be inferred. This also increases the number of sensors on the vehicle. Other techniques use sensor signals to indirectly detect wheel lifting qualitatively.

**[0008]** One example of a wheel lifting determination can be found in Ford patent U.S. 6,356,188 and U.S. patent application (Attorney Docket 202-0433/FGT-1683 PA), the disclosures of which are incorporated by reference herein. The system applies a change in torque to the wheels to determine wheel lift. The output from such a wheel lifting determination unit can be used qualitatively. This method is an active determination since the basis of the system relies on changing the torque of the wheels by the application of brakes or the like. In some situations it may be desirable to determine wheel lift without changing the torque of a wheel.

**[0009]** Due to the inevitable dead spots due to the vehicle configuration, wheel lift detection methods may be not able to identify all the conditions where four wheels are absolutely grounded in a timely and accurate fashion. For example, if the torques applied to the wheels have errors, if the vehicle reference computation has errors or there is not enough excitation in the torque provided, the wheel lift detection may provide

erroneous information or no information about the roll trending of the vehicle. Wheel lift information may also be safe-guarded by information regarding the vehicle roll angle information from the various sensors.

**[0010]** In certain driving conditions where the vehicle is moving with all four wheels contacting ground and the wheel lift detection does not detect the grounding condition, the roll information derived from the various sensors may be the sole information for identify vehicle roll trending. If in such driving cases, the vehicle experiences very large lateral acceleration and large roll rate, the grounded conditions might be replaced by erroneous lifting conditions. That is, those signals may predict that the vehicle is in a divergent roll event but the actual vehicle is not in a rolling event at all. Such cases include when the vehicle is driven on a mountain road, off-road or banked road, tire compression or an impact may cause a large normal load. The increased normal load causes a force component to be added to the lateral acceleration sensor output. Hence, a larger than 1 g lateral acceleration is obtained but the actual lateral acceleration of the vehicle projected along the road surface might be in 0.6 g range. An off-road driving condition may also be an off-camber driving condition. When a low speed vehicle is driven on an off-camber road with some hard tire compression or impact, the control system may be fooled to activate un-necessarily.

**[0011]** In order to reduce false activations, it would therefore be desirable to provide a rollover detection

system that sensitizes and desensitizes the roll control determination.

#### **SUMMARY**

**[0012]** The present invention improves the operation of a rollover stability control system (RSC) by controlling the safety device to provide improved performance. One way in which the improvement may be implemented is by controlling or improving the brake pressure prediction to improve the feel and performance time of the system.

**[0013]** In one embodiment, a method of controlling a safety system of a vehicle include determining a roll rate, determining a first control pressure in response to roll rate, determining a roll angle, and determining a second control pressure in response to the roll angle. The method further includes determining a final control pressure in response to the first control pressure and the second control pressure and controlling the safety system in response to the final control pressure.

**[0014]** One advantage of the invention is that some or all of the ways in which to improve the system set forth herein may be used alone or simultaneously to improve the rollover control system.

**[0015]** Other advantages and features of the present invention will become apparent when viewed in light of the detailed description of the preferred embodiment when taken in conjunction with the attached drawings and appended claims.

## **BRIEF DESCRIPTION OF THE DRAWINGS**

**[0016]** Figure 1 is a diagrammatic view of a vehicle with variable vectors and coordinator frames.

**[0017]** Figure 2 is an end view of an automotive vehicle on a bank with definitions of various angles including global roll angle, relative roll angle, wheel departure angle (WDA), road bank angle and body-to-road angle.

**[0018]** Figure 3A is an end view of an on-camber divergent vehicle tendency.

**[0019]** Figure 3B is an end view of an automotive vehicle in an off-camber divergent condition.

**[0020]** Figure 3C is an end view of a vehicle in an on-camber convergent condition.

**[0021]** Figure 3D is an end view of a vehicle in an off-camber convergent condition.

**[0022]** Figure 4A is a block diagram of a stability control system.

**[0023]** Figure 4B is a block diagram of the controller 26 used in the stability control system depicted in Figure 4A.

**[0024]** Figure 5 is a block diagrammatic view of the unit 27 depicted in Figure 4B, which is used for quantitatively and qualitatively determining rollover trend of a vehicle.

**[0025]** Figure 6 is a detailed block diagram of a transition controller of the present embodiment.

[0026] Figure 7 is flow chart of the operation of one embodiment of the transition controller.

[0027] Figure 8 is a detailed block diagram of a PID controller of the present embodiment.

[0028] Figure 9 is flow chart of the operation of one embodiment of the PID controller.

[0029] Figure 10 is plot of a proportional peak hold strategy.

#### **DETAILED DESCRIPTION**

[0030] In the following figures the same reference numerals will be used to identify the same components. The present teachings may be used in conjunction with a yaw control system or a rollover control system for an automotive vehicle. However, the present teachings may also be used with a deployment device such as airbag or roll bar.

[0031] Referring to Figure 1, an automotive vehicle 10 on a road surface 11 with a safety system is illustrated with the various forces and moments thereon. Vehicle 10 has front right and front left tires 12a and 12b and rear right tires and rear left tires 13a and 13b, respectively. The vehicle 10 may also have a number of different types of front steering systems 14a and rear steering systems 14b including having each of the front and rear wheels configured with a respective controllable actuator, the front and rear wheels having a conventional type system in which both of the front wheels are controlled together and both of the rear wheels are controlled together, a system having

conventional front steering and independently controllable rear steering for each of the wheels, or vice versa. Generally, the vehicle has a weight represented as  $Mg$  at the center of gravity of the vehicle, where  $g=9.8\text{ m/s}^2$  and  $M$  is the total mass of the vehicle.

**[0032]** As mentioned above, the system may also be used with active/semi-active suspension systems, anti-roll bar or other safety devices deployed or activated upon sensing predetermined dynamic conditions of the vehicle.

**[0033]** The sensing system 16 is part of a control system 18. The sensing system 16 may use a standard yaw stability control sensor set (including lateral acceleration sensor, yaw rate sensor, steering angle sensor and wheel speed sensor) together with a roll rate sensor and a longitudinal acceleration sensor. The various sensors will be further described below. The wheel speed sensors 20 are mounted at each corner of the vehicle, and the rest of the sensors of sensing system 16 may be mounted directly on the center of gravity of the vehicle body, along the directions  $x, y$  and  $z$  shown in Figure 1. As those skilled in the art will recognize, the frame from  $b_1, b_2$  and  $b_3$  is called a body frame 22, whose origin is located at the center of gravity of the car body, with the  $b_1$  corresponding to the  $x$  axis pointing forward,  $b_2$  corresponding to the  $y$  axis pointing off the driving side (to the left), and the  $b_3$  corresponding to the  $z$  axis pointing upward. The angular rates of the car body are denoted about their



respective axes as  $\omega_x$  for the roll rate,  $\omega_y$  for the pitch rate and  $\omega_z$  for the yaw rate. The calculations set forth herein may take place in an inertial frame 24 that may be derived from the body frame 22 as described below.

**[0034]** The angular rate sensors and the acceleration sensors are mounted on the vehicle car body along the body frame directions  $b_1, b_2$  and  $b_3$ , which are the  $x-y-z$  axes of the vehicle's sprung mass.

**[0035]** The longitudinal acceleration sensor 36 is mounted on the car body located at the center of gravity, with its sensing direction along  $b_1$ -axis, whose output is denoted as  $a_x$ . The lateral acceleration sensor 32 is mounted on the car body located at the center of gravity, with its sensing direction along  $b_2$ -axis, whose output is denoted as  $a_y$ .

**[0036]** The other frame used in the following discussion includes the road frame, as depicted in Figure 1. The road frame system  $r_1r_2r_3$  is fixed on the driven road surface, where the  $r_3$  axis is along the average road normal direction computed from the normal directions of the four-tire/road contact patches.

**[0037]** In the following discussion, the Euler angles of the body frame  $b_1b_2b_3$  with respect to the road frame  $r_1r_2r_3$  are denoted as  $\theta_{xr}, \theta_{yr}$  and  $\theta_{zr}$ , which are also called the relative Euler angles.

[0038] Referring now to Figure 2, the relationship of the various angles of the vehicle 10 relative to the road surface 11 is illustrated. One angle is a wheel departure angle  $\theta_{wda}$ , which is the angle from the axle or the wheel axis to the road surface 11. Also shown is a reference road bank angle  $\theta_{bank}$ , which is shown relative to the vehicle 10 on a road surface. The vehicle 10 has a vehicle body 10a and vehicle suspension 10b. The relative roll angle  $\theta_{xr}$  is the angle between the wheel axle and the body 10a. The global roll angle  $\theta_x$  is the angle between the horizontal plane (e.g., at sea level) and the vehicle body 10a.

[0039] Referring now to Figure 3A, vehicle 10 is illustrated in an on-camber divergent state. The on-camber divergent state refers to the vehicle having a greater than zero wheel departure angle, a greater than zero relative roll angle, and a moment represented by arrow 25 tending to increase the relative roll angle and the wheel departure angle. In this example, the bank angle is less than zero.

[0040] In Figure 3B, when the bank angle is greater than zero, the wheel departure angle is greater than zero, the relative roll angle is greater than zero and the moment is also to the right or increasing the relative roll angle and the wheel departure angle, the vehicle is in an off-camber divergent state.

[0041] Referring now to Figure 3C, a bank angle of less than zero, a wheel departure angle greater than zero, and a relative roll angle greater than zero is

shown with a roll moment 25 acting to the left. Thus, the vehicle is in an on-camber convergent state. That is, the convergent state refers to the vehicle tending towards not overturning.

**[0042]** Referring now to Figure 3D, when the bank angle is greater than 0, the wheel departure angle is greater than zero, and the relative roll angle is greater than zero and the roll moment is tending to the left, the vehicle is in an off-camber convergent state. That is, the vehicle is tending toward not rolling over.

**[0043]** Referring now to Figure 4A, one embodiment of a roll stability control system 18 is illustrated in further detail having a controller 26 used for receiving information from a number of sensors which may include a yaw rate sensor 28, a speed sensor 20, a lateral acceleration sensor 32, a roll rate sensor 34, a steering angle sensor (hand wheel position) 35, a longitudinal acceleration sensor 36, and steering angle position sensor 37.

**[0044]** In one embodiment, the sensors are located at the center of gravity of the vehicle. Those skilled in the art will recognize that the sensors may also be located off the center of gravity and translated equivalently thereto.

**[0045]** Lateral acceleration, roll orientation and speed may be obtained using a global positioning system (GPS). Based upon inputs from the sensors, controller 26 may control a safety device 38. Depending on the desired sensitivity of the system and various other factors, not all the sensors 20, 28, 32, 34, 35, 36, and

37, or various combinations of the sensors, may be used in a commercial embodiment. Safety device 38 may control an airbag 40, an active braking system 41, an active front steering system 42, an active rear steering system 43, an active suspension system 44, and an active anti-roll bar system 45, or combinations thereof. Each of the systems 40-45 may have their own controllers for activating each one. As mentioned above, the safety system 38 may be at least the active braking system 41.

**[0046]** Roll rate sensor 34 may sense the roll condition of the vehicle based on sensing the height of one or more points on the vehicle relative to the road surface. Sensors that may be used to achieve this include a radar-based proximity sensor, a laser-based proximity sensor and a sonar-based proximity sensor.

**[0047]** Roll rate sensor 34 may also sense the roll condition based on sensing the linear or rotational relative displacement or displacement velocity of one or more of the suspension chassis components which may include a linear height or travel sensor, a rotary height or travel sensor, a wheel speed sensor used to look for a change in velocity, a steering wheel position sensor, a steering wheel velocity sensor and a driver heading command input from an electronic component that may include steer by wire using a hand wheel or joy stick.

**[0048]** The roll condition may also be sensed by sensing the force or torque associated with the loading condition of one or more suspension or chassis components including a pressure transducer in active air suspension, a shock absorber sensor such as a load cell,

a strain gauge, the steering system absolute or relative motor load, the steering system pressure of the hydraulic lines, a tire lateral force sensor or sensors, a longitudinal tire force sensor, a vertical tire force sensor or a tire sidewall torsion sensor.

**[0049]** The roll condition of the vehicle may also be established by one or more of the following translational or rotational positions, velocities or accelerations of the vehicle including a roll gyro, the roll rate sensor 34, the yaw rate sensor 28, the lateral acceleration sensor 32, a vertical acceleration sensor, a vehicle longitudinal acceleration sensor, lateral or vertical speed sensor including a wheel-based speed sensor, a radar-based speed sensor, a sonar-based speed sensor, a laser-based speed sensor or an optical-based speed sensor.

**[0050]** Based on the inputs from sensors 20, 28, 32, 34, 35, 36, 37, controller 26 determines a roll condition and controls any one or more of the safety devices 40-45.

**[0051]** Speed sensor 20 may be one of a variety of speed sensors known to those skilled in the art. For example, a suitable speed sensor 20 may include a sensor at every wheel that is averaged by controller 26. The controller 26 translates the wheel speeds into the speed of the vehicle. Yaw rate, steering angle, wheel speed and possibly a slip angle estimate at each wheel may be translated back to the speed of the vehicle at the center of gravity. Various other algorithms are known to those skilled in the art. For example, if speed is determined while speeding up or braking around a corner,

the lowest or highest wheel speed may not be used because of its error. Also, a transmission sensor may be used to determine vehicle speed.

**[0052]** Referring now to Figures 4A and 4B, controller 26 is illustrated in further detail. There are two major functions in controller 26: the rollover trend determination, which is called a sensor fusion unit 27A, and the feedback control command unit 27B. The sensor fusion unit 27A can be further decomposed as a wheel lift detector 50, a transition detector 52 and a vehicle roll angle calculator 66.

**[0053]** Referring now to Figure 5, the sensor fusion unit 27A is illustrated in further detail. The sensor fusion unit 27A receives the various sensor signals, 20, 28, 32, 34, 35, 36, 37 and integrates all the sensor signals with the calculated signals to generate signals suitable for roll stability control algorithms. From the various sensor signals wheel lift detection may be determined by the wheel lift detector 50. Wheel lift detector 50 includes both active wheel lift detection and active wheel lift detection, and wheel grounding condition detection. Wheel lift detector is described in co-pending U.S. provisional application serial number 60/400,375 (Attorney Docket 202-0433/FGT-1683PRV) filed August 1, 2002, and U.S. patent application (Attorney Docket 202-0433/FGT-1683PA) which are incorporated by reference herein. The modules described below may be implemented in hardware or software in a general purpose computer (microprocessor). From the wheel lift detection module 50, a determination of whether each wheel is absolutely grounded, possibly grounded,

possibly lifted, or absolutely lifted may be determined. Transition detection module 52 is used to detect whether the vehicle is experiencing aggressive maneuver due to sudden steering wheel inputs from the driver. The sensors may also be used to determine a relative roll angle in relative roll angle module 54. Relative roll angle may be determined in many ways. One way is to use the roll acceleration module 58 in conjunction with the lateral acceleration sensor. As described above, the relative roll angle may be determined from the roll conditions described above.

**[0054]** The various sensor signals may also be used to determine a relative pitch angle in relative pitch angle module 56 and a roll acceleration in roll acceleration module 58. The outputs of the wheel lift detection module 50, the transition detection module 52, and the relative roll angle module 54 are used to determine a wheel departure angle in wheel departure angle module 60. Various sensor signals and the relative pitch angle in relative pitch angle module 56 are used to determine a relative velocity total in module 62. The road reference bank angle block 64 determines the bank angle. The relative pitch angle, the roll acceleration, and various other sensor signals as described below are used to determine the road reference bank angle. Other inputs may include a roll stability control event (RSC) and/or the presence of a recent yaw stability control event, and the wheel lifting and/or grounding flags.

**[0055]** The global roll angle of the vehicle is determined in global roll angle module 66. The relative roll angle, the wheel departure angle, and the roll

velocity total blocks are all inputs to the global roll angle total module 66. The global roll angle total block determines the global roll angle  $\theta_x$ . An output module 68 receives the global roll angle total module 66 and the road reference bank angle from the road reference bank angle module 64. A roll signal for control is developed in roll signal module 70. The roll signal for control is illustrated as arrow 72. A sensitizing and desensitizing module 74 may also be included in the output module 68 to adjust the roll signal for control.

**[0056]** In the reference road bank angle module 64, the reference bank angle estimate is calculated. The objective of the reference bank estimate is to track a robust but rough indication of the road bank angle experienced during driving in both stable and highly dynamic situations, and which is in favor for roll stability control. That is, this reference bank angle is adjusted based on the vehicle driving condition and the vehicle roll condition. Most importantly, when compared to the global roll estimate, it is intended to capture the occurrence and physical magnitude of a divergent roll condition (two wheel lift) should it occur. This signal is intended to be used as a comparator against the global roll estimate for calculating the error signal which is fed back to roll stability controller 26.

**[0057]** In parallel with the above a transition controller 76 may implemented as will be further described below. The roll signal for control 72 may be used as an input to a proportional-integral-derivative



controller 78. The terminology for the PID controller 78 refers to its functions. However, the function of double derivative may be added and a function such as integral may be used. For clarity the PID controller will be used for the controller even if all of the function proportional, integral or derivative functions are not used or if the double derivative is used. In parallel to the process above, a transitional controller may also be used. The transitional controller 78. One embodiment for example includes just the proportional and derivative functions.

**[0058]** The outputs of controller 76 and the controller 78 are provided to an arbitration module 80, which ultimately controls the safety system. In the present example, the safety system is a hydraulic safety system such as a rollover control system using brakes. The arbitration module 80 may, for example, simply choose the highest brake pressure requested from the transition controller and the PID controller. Of course, a weighting may also be performed.

**[0059]** Referring now to Figure 6, the operation of the transition controller 76 is described in further detail. In this module, RSC control for the transitional portion of a dynamic maneuver is performed. The transitional portion is the region in which the outputs of the sensors are still linear. That is, none are saturated. As briefly described below, the transition controller may perform all or some of the following.

**[0060]** Caliper pre-charge functionality. During control interventions requiring maximum pressure build

rates, significant delays in pressure builds occur due to the relatively large volume of fluid required to establish caliper pressure in the lower pressure range. The higher volume consumption is due to the air gap between the rotor and linings, as well as the high effective compliance of the brake system at low pressures. The transition controller 76 includes pre-charging functionality to mitigate initial delays in establishing caliper pressure by applying low levels of pressure when steering activity suggests an RSC event is eminent.

**[0061]** Yaw Damping. The under-damped nature of the yaw dynamics of the vehicle can result in yaw rate overshoot in transitional maneuvers. Excessive yaw rate overshoot in limit maneuvers results in excessive side slip angles which can result in lateral forces that significantly exceed the steady state cornering capacity of the vehicle and can significantly reduce the roll stability margin of the vehicle. As a preventative measure, yaw damping may be provided to minimize the occurrence of excessive lateral forces and vehicle side slip angles that might expose the vehicle to excessive lateral forces and tripping mechanisms. The phase of the brake interventions resulting from this module introduces significant yaw damping during aggressive maneuvers. A goal is to provide as much yaw damping as possible without inhibiting the responsiveness of the vehicle or becoming intrusive.

**[0062]** Roll Damping. For sufficiently aggressive transitional maneuvers, the roll momentum can result in a lifting of the center of gravity of the vehicle at the

end of the transition and may result in excessive wheel lift. It is an objective of this module to introduce effective roll damping before the occurrence of wheel lift by rounding off the build up lateral force when needed as they approach their peak levels in the final phase of the transition.

**[0063]** Pressure Build Prediction. In addition to the caliper pre-charge functionality, pressure build prediction and actuator delay compensation have been introduced in this module. Limitations in pressure build rates are compensated for by projecting forward when a pre-determined pressure level is likely to be requested, based on the relative roll angle, roll rate, roll acceleration, and estimated caliper pressure. Pressure is built during the transition so that the desired peak pressure can be achieved when it is needed to reduce the effects of limited pressure build rates.

**[0064]** Feed forward control. In this module, steering information, front side slip angle information and relative roll information is used to achieve a feed forward component of control from the point of view of the PID controller 78. Feed forward information is used to build significant pressure where needed to reduce the demands on the PID controller 78 for a given RSC event and to extend the functional envelop of the RSC system. For mild and moderate events, stability performance is achieved with lower levels of PID intervention. In conjunction with feed-forward control, PID control is able to handle more extreme maneuvers than would be possible without feed-forward control for two primary reasons. First, by the time the PID controller 78 has

requested intervention, the vehicle will be in a more stable condition due to the feed forward control. Second, when the PID controller 78 does request pressure, significant pressure will already exist in the caliper, allowing the PID controller to achieve the required pressure much more quickly.

**[0065]** Gradient Control. Proportional plus Derivative control is implemented for roll angle and front linear side slip angle in the PID controller 78. For the PID controller 78, relative roll is used for the proportional term, and roll velocity is used for the derivative term. Because relative roll is used, this controller is robust to integration errors that might occur in the total roll estimate. As a result, it is possible to make the transition controller 76 more sensitive than the PID controller without the adverse consequences of integrator drift in the state estimates.

**[0066]** Establish initial conditions for PID controller 78. Because this module is designed to lead the PID control intervention in a given maneuver, PID control can be initiated at significantly higher pressure, requiring less error to achieve the critical pressure required to stabilize the vehicle.

**[0067]** Progressive Pressure Build. The transition controller 76 builds and decreases pressure with a minimum amount of jerk. This supports smooth transparent interventions and reduces the potential for exciting pitch dynamics in the vehicle.

**[0068]** Transparency. The transition controller 76 may also builds pressure in parallel with the transition

in lateral acceleration. Because the longitudinal acceleration is building at the same time as the reversal in lateral acceleration, the pitch motion is more in phase with the roll motion, resulting in a coherent body motion. Additionally, because the build up of longitudinal acceleration is much smaller than the change in lateral acceleration, the brake intervention becomes overshadowed by the lateral jerk.

**[0069]** Referring now to Figure 6, the transition controller 76 includes the following inputs:

a filtered steering angle rate (FILTERED\_STEER\_ANGLE\_RATE) input 90, a final wheel lift status (FINAL\_WHEEL\_LIFT\_STATUS) 92, a filtered lateral acceleration (FLT\_LAT\_ACC) input 94, a filtered roll rate (FLT\_ROLL\_RATE) input 96, a filtered yaw rate (FLT\_YAW\_RATE) signal 98, a Get\_ayc\_reverse\_movement input 100, a Get\_rsc\_disabled input 102, a PID\_INCREASE\_REQUESTED\_PRESSURE[FL] input 104, a PID\_INCREASE\_REQUESTED\_PRESSURE[FR] input 106, a PID\_STBLZ\_PRES[FL] input 108, a PID\_STBLZ\_PRES[FR] input 110, a RECENT\_LEFT\_WHEEL\_LIFT input 112, a RECENT\_PID\_ACTIVATION input 114, a RECENT\_PID\_CONTROL input 116, a RECENT\_RIGHT\_WHEEL\_LIFT input 118, a REF\_VELOCITY\_YC input 119, a REL\_ROLL\_ANGLE input 120, a RSC\_Disable\_after\_sensor\_self\_test input 122, an RSC\_DISABLED input 124, a SLOW\_FLT\_YAW\_ACC input 126, a Status\_first\_run input 128, a STEERING\_WHEEL\_ANGLE input 130, a WDA\_LIFT\_SUSPECTED input 132, Z1\_REL\_ROLL\_ANGLE input 134, a ss\_dpss\_YAW\_ACCELERATION2 input 136, a AYC\_SLIP\_ANGLE\_RATE\_RA input 138, a WHEEL\_DEPARTURE\_ANGLE input 140, and a Predicted\_AY\_from\_SWA input 144.

[0070] The transition controller 78 has the following outputs FLPrechargeActive (flag) output 150, a FLPrechargePress (bar) output 152, a FRPrechargeActive (flag) output 154, a FRPrechargePress (bar) output 156, an INST\_BANK\_ANGLE\_EST output 158, a LEFT\_TO\_RIGHT\_TRANSITION (flag) output 160, and a RIGHT\_TO\_LEFT\_TRANSITION (flag) output 162.

[0071] The transition controller 78 also has predefined calibratable parameter that have the following definitions:

scale\_LNR\_REAR\_LATA\_FILTER: In the present example a value of 1024.0 unitless was used.

value\_LNR\_REAR\_LATA\_LOW\_PASS\_FILTER\_FREQ: In the present example a value of 10.0 Hz was used.

$$p\_ALPHA\_LNR\_REAR\_LATA\_FILTER\_COEFF: (LOOP\_TIME * \\ value\_LNR\_REAR\_LATA\_LOW\_PASS\_FILTER\_FREQ * \\ scale\_LNR\_REAR\_LATA\_FILTER )$$

$$p\_BETA\_LNR\_REAR\_LATA\_FILTER\_COEFF: ((1.0 - (LOOP\_TIME * \\ value\_LNR\_REAR\_LATA\_LOW\_PASS\_FILTER\_FREQ)) * \\ scale\_LNR\_REAR\_LATA\_FILTER )$$

BASE\_MATCHING\_PRESSURE: In the present example a value of -177.0 bar was used. The matching pressure for the precharge prediction at a mu level of zero.

MATCHING\_PRESSURE\_MU\_GAIN: In the present example a value of 3000 bar/g was used. The gain with which the matching pressure is increased as mu is increased.

EXPECTED\_SWA\_LEAD\_TIME\_MU\_GAIN: The gain for determining how far in advance the predicted steering wheel angle sign crossing will cause a positive or negative SWA expected flag. In the following example, values 0.4- 0.5 sec/g were used.

SpeedDpndntAyChng (linear interpolation function of vehicle speed): The speed dependent threshold for determining if the driver is requesting a rapid change in lateral acceleration. The following table illustrates the numbers used in the present example.

V (mps)	SpeedDpndntAyChng (mps <sup>2</sup> /sec)
0	33
17.9	33
22.4	30
26.79	25
31.3	16
83	11

RECENT\_RAPID\_AY\_TIME: In the present example a value of 150 loops was used. The time in loops for which the recent rapid ay flag is held high after the rate of change of the lateral acceleration requested by the driver exceeds the threshold.

LARGE\_ROUGH\_MU: In the present example a value of 0.7 g's was used. Large rough mu threshold for determining significant dynamics.

RECENT\_LARGE\_ROUGH\_MU\_TIME: In the present example a value of 150 loops was used. The time in loops for which the recent large rough mu flag is held high after the rough mu estimate exceeds a predetermined threshold.

KP\_REL\_ROLL: In the present example a value of 8.0 bar/deg was used. Proportional pressure gain for relative roll feedback.

KD\_REL\_ROLL: In the present example a value of 2.0 bar/deg/s was used. Derivative pressure gain for relative roll feedback.

REL\_ROLL\_DB: In the present example a value of 5.4 deg was used. Deadband for proportional feedback on relative roll.

REL\_ROLL\_PD\_PRESS\_FILT: In the present example a value of 1.0/5.0 unitless was used. Filter coefficient used in filtering the relative roll Proportional plus derivative feedback requested pressure.

KP\_SS\_LIN\_FRONT: In the present example a value of 10.0 bar/deg was used. Proportional pressure gain for front linear side slip angle feedback.



KD\_SS\_LIN\_FRONT: In the present example a value of 2.5 bar/deg/s was used. Derivative pressure gain for front linear side slip angle feedback.

FT\_CRNRG\_CMPLNCE Front cornering compliance: In the present example, values between 6 and 8.18 deg/g were used.

RR\_CRNRG\_CMPLNCE Rear cornering compliance: In the present example, values between 3 and 5 deg/g were used.

SS\_LIN\_FRONT\_DB: In the present example a value of 8.2 deg/g was used. Deadband for proportional feedback on front linear side slip angle feedback.

SS\_LIN\_PD\_PRESS\_FILT: In the present example a value of 1.0/16.0 unitless was used. Filter coefficient used in filtering the front linear side slip angle proportional plus derivative feedback requested pressure.

MAX\_PRECHARGE\_TIME: In the present example a value of 70 loops was used. Time in loops after which under certain conditions, the precharge pressure is forced down at the maximum rate.

PRECHARGE\_BUILD\_TIME: In the present example a value of 20 loops was used. Time in loops before which under certain conditions, the precharge pressure is ramped down at the target build rate.

TARGET\_BUILD\_RATE: In the present example a value of 100.0 bar/sec was used. The target build rate for the precharge strategy while in prediction mode.

LARGE\_DYNAMICS\_BUILD\_RATE: In the present example a value of 250.0 bar/sec was used. The pressure build rate for the precharge strategy when either the relative roll or front linear side slip angle PD pressures exceed the current precharge pressure on a given corner of the vehicle.

VELOCITY\_MIN\_10: In the present example a value of 10.0 mps was used. Minimum velocity below which the precharge strategy will not be enabled.

ROUGH\_MU\_HOLD\_TIME: In the present example a value of 150 counts was used. The time in loops for which the rough mu estimate will be held high after an increase in value.

ROUGH\_MU\_RAMP\_DOWN\_RATE: In the present example a value of 0.2g's/sec was used. The rate at which the rough mu estimate will be ramped down after the rough mu hold time is exceeded.

BASE\_PRESSURE\_MU\_GAIN: In the present example a value of 15.0 bar/g was used. The gain for determining the precharge pressure from rough mu, due to rapid steering input.

RATE\_LIMIT\_dAY\_FROM\_SWA: In the present example, values between 70 mps<sup>2</sup>/s/s; and 140 mps<sup>2</sup>/s/s were used. The rate limit for the decrease in value of the rate limited signal representing the rate of change of lateral acceleration being requested by the driver.

MIN\_FL\_REL\_ROLL\_PD\_PRESS: In the present example a value of -320 bar was used. Lower bound for relative roll proportional/derivative pressure.

MAX\_FL\_REL\_ROLL\_PD\_PRESS: In the present example a value of -320 bar was used. Upper bound for relative roll proportional/derivative pressure.

FAST\_PRESS\_REDUCTION\_RATE: In the present example a value of 200 bar/s was used. The fast pressure reduction rate for the precharge strategy while in prediction mode.

AVG\_Steering\_Ratio: In the present example a value of 16 deg/deg was used. The ratio of steering to wheel.

Rel\_Roll\_Vel\_FLT\_Coeff: In the present example a value of (16) 1/16<sup>th</sup> filter coefficient was used in filtering relative roll velocity.

MIN\_REF\_VELOCITY\_YC: In the present example a value of 0.01 kph was used to prevent divide by 0 when the reference velocity is too small.

lin\_frnt\_vel\_filter\_coeff: In the present example a value of (16.0)  $1/16^{\text{th}}$  filter coefficient was used in filtering linear front velocity.

max\_RoughMU: In the present example a value of 1.0 was used. Upper bound for Rough MU upper and lower

min\_RoughMU: In the present example a value of 0.1 was used. Lower bound for Rough MU upper and lower

MAX\_THRESHOLD\_FOR\_PREDICTED\_POS\_SWA: In the present example a value of -90 degrees was used. SWA threshold for predicting crossover to the left.

MIN\_THRESHOLD\_FOR\_PREDICTED\_POS\_SWA: In the present example a value of 90 degrees was used. SWA threshold for predicting crossover to the right.

MIN\_POS\_SWA\_THRESHOLD: In the present example a value of 0.1 degree was used. The SWA threshold for predicting crossover to the left.

MAX\_POS\_SWA\_THRESHOLD: In the present example a value of -0.1 degree was used. The SWA threshold for predicting crossover to the right.

FRNT\_SLIP\_ANGLE\_THRESHOLD\_POS\_SIG\_DYN: In the present example a value of 0.4 Force threshold was used in setting Significant\_Dynamics flag.

VELOCITY\_MIN\_TO\_EXIT\_7: In the present example a value of 7.0 m/s was used. Lower bound on longitudinal velocity for setting ENABLE\_RSC\_PRECHARGE flag.

MIN\_SLIP\_ANGLE\_FOR\_PRCHG\_ACT: In the present example a value of 0.1 degrees was used. Minimum slip angle for precharge\_active flags.

TRANSITION\_AY\_THRESHOLD: In the present example a value of 0.5 G's was used. Lateral acceleration threshold to initiate moderate ay counters.

TRANSITION\_HOLD\_TIME: In the present example a value of 100 loops was used. Initial value for moderate ay counters.

VHCL\_MASS: In the present example a value of 3033.0 kg was used. Vehicle mass.

CG\_TO\_RR\_AXLE: In the present example a value of 1.471 m was used. Distance from vehicle center of gravity to rear axle.

CG\_TO\_FRT\_AXLE: In the present example a value of 1.546 m was used. Distance from vehicle center of gravity to front axle.

AY\_SNSR\_TO\_RR\_AXLE: In the present example a value of -1.791 m was used. Distance from ay sensor to rear axle.

WHEEL\_BASE (CG\_TO\_FRT\_AXLE + CG\_TO\_RR\_AXLE)

AY\_TO.CG (CG\_TO\_RR\_AXLE + AY\_SNSR\_TO\_RR\_AXLE)

YAW\_MMT\_INERTIA: In the present example a value of 6303.5 kg-m/s<sup>2</sup> was used. Moment of inertia about the Z axis.

G\_TO\_MPS2: In the present example a value of 9.81 m/s<sup>2</sup>/G was used. Convert from G's to m/s<sup>2</sup>.

value\_MAX\_ANGLE (45.0 degrees): Upper limit for the estimate of the instantaneous bank angle

MAX\_ANGLE (value\_MAX\_ANGLE/deg\_RES\_ROLL\_ANGLE)

ss\_RES\_dAYFromSWA: In the present example a value of 0.004375 was used. Scaling for dAYFromSWA.

MIN\_dAY\_FROM\_SWA: In the present example a value of -100.00/ss\_RES\_dAYFromSWA was used. Lower limit for dAYFromSWA.

MAX\_dAY\_FROM\_SWA: In the present example a value of 100.00/ss\_RES\_dAYFromSWA was used. Upper limit for dAYFromSWA.

LARGE\_SWA\_RATE\_THRESH: In the present example a value of 700 deg/s was used. Threshold to reset recent large swa rate timers.

RECENT\_WHEEL\_LIFT\_TIME: In the present example a value of 1/looptime was used. Initial value for recent wheel lift timers.

rpss\_RES\_YAW\_ACC (dpss\_RES\_YAW\_ACC \*pc\_DEG\_TO\_RAD):  
Scaling for yaw acceleration in radians.

MIN\_dAY\_FROM\_YAW\_ACC: In the present example a value of -100.0/ss\_RES\_dAYFromSWA was used. Lower limit for ssDayFromYawAcc.

MAX\_dAY\_FROM\_YAW\_ACC: In the present example a value of 100.00/ss\_RES\_dAYFromSWA was used. Upper limit for ssDayFromYawAcc.

p\_dpg\_RR\_CRNRG\_CMPLNCE: In the present example a value of 4.68 deg/g U228 was used. Rear cornering compliance.

dps\_RES\_BETA\_PRIME(rps\_RES\_YAW\_RATE\*pc\_RAD\_TO\_DEG):  
Scaling for ssDayFromBetaPRA.

MIN\_dAY\_FROM\_BETAPRA: In the present example a value of -140.0/ss\_RES\_dAYFromSWA was used. Lower limit for ssDayFromBetaPRA.

MAX\_dAY\_FROM\_BETAPRA: In the present example a value of 140.0/ss\_RES\_dAYFromSWA was used. Upper limit for ssDayFromBetaPRA.

YawAcc2SWARate\_x\_tab: In the present example a value of {0, 30, 50, 70, 203} MPH was used. X interpolation table for ssYawAcc2SWARateFraction.

YawAcc2SWARate\_y\_tab: In the present example a value of {0.66, 0.66, 0.60, 0.50, 0.50} fraction was used. Y interpolation table for ssYawAcc2SWARateFraction.

YawAcc2SWARate\_len: In the present example a value of 5 unitless was used. The number of entries in the interpolation tables for ssYawAcc2SWARateFraction.

BetaDotRA2SWARate\_x\_tab: In the present example a value of {0, 30, 50, 203} MPH was used. X interpolation table for ssBetaDotRA2SWARateFraction.

BetaDotRA2SWARate\_y\_tab: In the present example a value of {0.35, 0.35, 0.50, 0.50} fraction was used. Y interpolation table for ssBetaDotRA2SWARateFraction.

BetaDotRA2SWARate\_len: In the present example a value of 4 unitless was used. The number of entries in the interpolation tables for ssBetaDotRA2SWARateFraction.

Build\_rate\_x\_tab: In the present example a value of {0, 30, 50, max} mps2/sec was used. X lookup table for build rate.

Build\_rate\_y\_tab: In the present example a value of {100, 100, 250, 250} bar/sec was used. Y lookup table for build rate.

Build\_rate\_tab\_len: In the present example a value of 4 unitless was used. Length of build rate lookup tables for interpolation.

FRNT\_SLIP\_ANGLE\_THRESHOLD\_POS\_SIG\_DYN: In the present example a value of 0.4 G was used. Threshold for Front linear slip angle.

MIN\_MATCH\_PRESS\_2\_PD\_PRESS\_DELTA: In the present example a value of 1.0 bar was used. Lower limit on PDRelRollMatchPressDelta.

MIN\_INST\_BUILD\_RATE: In the present example a value of 100.0 bar was used. Minimum for InstRelRollBuildRate.

MAX\_INST\_BUILD\_RATE: In the present example a value of 300.0 bar was used. Maximum for InstRelRollBuildRate.

TARGET\_DUMP\_RATE: In the present example a value of 150 bar/s was used. Rate for reducing pressure.

UPPER\_BOUND\_ON\_TIME\_FOR\_DETECTION: In the present example a value of 0.400 was used. Upper bound in time for increasing swa detection.

YAW\_RATE\_PD\_PRESS\_FILT: In the present example a value of 1.0/16.0 was used. Corresponds to roughly 2 Hz cutoff.

AY\_FROM\_SWA\_THRESH\_FOR\_DIRECTION\_CHANGE: In the present example a value of 0.5 g was used. Sign dependent threshold used in the first set of conditions that must exist to disable the yaw rate PD controller from increasing final requested pressure -- when the magnitude of predicted lateral acceleration from steering wheel angle drops below this threshold.

TIME\_FOR\_DETECTING\_SWA\_DIRECTION\_CHANGE: In the present example a value of .020 sec was used. Minimum contiguous time used to indicate a steering wheel direction change - used as the second set of conditions to disable the yaw rate PD controller from increasing final requested pressure.

SWA\_VEL\_FOR\_DETECTING\_SWA\_DIRECTION\_CHANGE: In the present example a value of 400 deg/s was used. Sign dependent steering velocity threshold which is multiplied by parameter TIME\_FOR\_DETECTING\_SWA\_DIRECTION\_CHANGE to define the minimum change required in steering wheel angle magnitude, defining the third condition that must be met before disabling the yaw rate PD controller.

Rsc\_kp\_yaw\_rate Yaw Rate Controller Kp

Rsc\_kd\_yaw\_rate Yaw Rate Controller Kd

**[0072]** Referring now to Figure 7, a flowchart of one embodiment of the operation of the transition controller 76 is illustrated. It should be noted that not all steps may be included in an actual embodiment. Further, the parameters and constants are set forth by way of example and are not meant to be limiting. In step 180, the various sensors described above are read. In step 182 various inputs from other sensors are obtained. These inputs correspond to inputs 90-142 of Figure 6. In step 184 the local variables described above are

obtained. These local variables include the lateral force, slip angle and lateral velocity at the front axle. These are determined as follows:

```
FrontLatForce=((VHCL_MASS*CG_TO_RR_AXLE/WHEEL_BASE)
*FLT_LAT_ACC)
+(SLOW_FLT_YAW_ACC*((VHCL_MASS*CG_TO_RR_AXLE
*AY_TO_CG)
+YAW_MMT_INERTIA)/(WHEEL_BASE);
```

```
FrontLnrSlipAngle=-FrontLatForce*(FT_CRNRG_CMPLNCE
*WHEEL_BASE)/(VHCL_MASS*CG_TO_RR_AXLE*G_TO
_MPS2);
```

```
CurrentFrontLnrLatVel=FrontLnrSlipAngle*REF
_VELOCITY_YC;
```

**[0073]** Also the lateral force, slip angle and lateral velocity at the rear axle are determined as follows:

```
RearLatAcc=FLT_LAT_ACC+(SLOW_FLT_YAW_ACC*AY_SNSR_TO
_RR_AXLE);
```

```
RearLatForce=(VHCL_MASS*(CG_TO_FRT_AXLE/WHEEL_BASE)
*FLT_LAT_ACC)+(SLOW_FLT_YAW_ACC*((VHCL_MASS*CG
_TO_FRT_AXLE*AY_TO_CG)-YAW_MMT_INERTIA)
/WHEEL_BASE;
```

```
RearLnrSlipAngle=(-1)*RearLatForce*(RR_CRNRG
_CMPLNCE*WHEEL_BASE)/(VHCL_MASS*CG_TO_FRT_AXLE
*G_TO_MPS2);
```

```
CurrentRearLnrLatVel=RearLnrSlipAngle*REF_VELOCITY
_YC;
```

**[0074]** In step 186 the component of the lateral acceleration at the rear axle due to the change in linear lateral velocity at the rear axle is calculated as follows:

```
RearLnrLatAcc=(CurrentRearLnrLatVel-REAR_LNR
_LAT_VEL)/LOOP_TIME;
```



```

FltRearLnrLatA=ALPHA_LNR_REAR_LATA_FILTER_COEF
F*RearLnrLatAcc+BETA_LNR_REAR_LATA_FILTER
_COEFF*REAR_LNR_LAT_ACC;

```

**[0075]** In step 188 the instantaneous bank angle based on rear axle information is determined as follows:

```

SineOfInstBnkAngle=((RearLatAcc-FltRearLnrLatA)
-(FLT_YAW_RATE*REF_VELOCITY_YC))*MPS2_TO_G;

//Limit magnitude of Bank.Angle to 45 degrees if
(SineOfInstBnkAngle<=-SCALED_ASIN_INPUT_LIMIT)
INST_BANK_ANGLE_EST = -1.0 * MAX_ANGLE;
else if (SineOfInstBnkAngle>=SCALED_ASIN_INPUT
LIMIT)INST_BANK_ANGLE_EST= MAX_ANGLE;
else INST_BANK_ANGLE_EST=COMPUTE_ARC_SINE_OF_INPUT
(SineOfInstBnkAngle);

```

**[0076]** In step 190 the rate of change of requested lateral acceleration based on steering velocity is determined. A product of the steering velocity multiplied by the steering gain of the vehicle is obtained, where the steering gain is defined as the ratio of the lateral acceleration to steering wheel angle in a steady state turn. This gain is expressed as a function of speed. The product is passed through a low pass filter to achieve the resultant rate of change of lateral acceleration being requested by the driver. This signal is used to help define the strength of the transition being requested by the driver. This determined in the flowing:

```

temp_For_SWA=REF_VELOCITY_YC*REF_VELOCITY_YC/
(RAD_TO_DEG*WHEEL_BASE*G_TO_MPS2);

K_SWA_TO_AY=temp_For_SWA*G_TO_MPS2/(1+temp_For_SWA
*(FT_CRNRG_CMPLNCE-RR_CRNRG_CMPLNCE));

dAYFromSWA=LIMIT(FILTERED_STEER_ANGLE_RATE*K_S
WA_TO_AY/AVG_Steering_Ratio, min_dAY_from_SWA,
max_dAY_from_SWA);

```

[0077] In step 192 a rate limited lateral acceleration from steering wheel angle dAYfromSWA is determined. The delay factor in the response of the power steering system may cause the system to have to catch-up to the pressure desired by the vehicle operator. For very aggressive steering inputs, power steering catch-up can cause a significant reduction in steering velocity for up to 200ms. This can cause an interruption in pressure build when it is needed most. To prevent this, rate limited versions of dAYfromSWA are implemented for both the positive and negative directions. Each signal is allowed to increase in value as rapidly as the input signal. However, the decrease in value is rate limited by RATE\_LIMIT\_dAY\_FROM\_SWA. Both signals are limited to positive values and represent the magnitude of the rate of change in the positive and negative direction respectively. This is determined as follows:

```

if(dAYFromSWA>POSRateLimiteddAYFromSWA) POSRate
    LimiteddAYFromSWA=dAYFromSWA;
else
{
    POSRateLimiteddAYFromSWA-=RATE_LIMIT_dAY
        _FROM_SWA*LOOPTIME;
    if(POSRateLimiteddAYFromSWA<dAYFromSWA) POSRate
        LimiteddAYFromSWA=dAYFromSWA;
}
POSRateLimiteddAYFromSWA=MAX(0, POSRateLimited
    dAYFromSWA);

if(-
dAYFromSWA>NEGRateLimiteddAYFromSWA) NEGRateLimited
    dAYFromSWA=-dAYFromSWA;
else
{
    NEGRateLimiteddAYFromSWA-=RATE_LIMIT_dAY
        _FROM_SWA*LOOPTIME;
    if(NEGRateLimiteddAYFromSWA<(-1*dAYFromSWA))

```

```

        NEGRateLimiteddAYFromSWA=-1*dAYFromSWA;
    }
    NEGRateLimiteddAYFromSWA=MAX(0,NEGRateLimiteddAY
    FromSWA);

```

**[0078]** In step 194 a change in lateral acceleration from yaw acceleration dAYfromYawAcc and a change in lateral acceleration from the rate of change of the side slip at the rear axle in dAYfromBetaPRA is determined. Each signal is allowed to increase in value as rapidly as the input signal. These signals are scaled to provide a similar profile to DayFromSWA. Final max and min DayDt's are calculated by comparing dAYfromYawAccScldToSWA, dAYfromBetaPRAScldToSWA, and dAYFromSWA. These DayDt's can be positive or negative. Max and min rate limited DayDt's are calculated by comparing dAYfromYawAccScldToSWA, dAYfromBetaPRAScldToSWA, and rate limited DayFromSWA's. These rate limited DayDt's are always positive. They are calculated as follows:

```

ssDayFromYawAcc=LIMIT((ss_dpss_YAW_ACCELERATION2
    *ss_mps_REF_VELOCITY_YC/(ss_RES_dAYFromSWA/
    (rpss_RES_YAW_ACC*mps_RES_REF_VELOCITY_YC))),
    MIN_dAY_FROM_YAW_ACC, MAX_dAY_FROM_YAW_ACC);

ssDayFromBetaPRA =LIMIT((AYC_SLIP_ANGLE_RATE_RA
    *(1024.0)/ ((ss_RES_DayFromSWA*p_dpg_RR
    _CRNRG_CMPLNCE*1024.0)/(dps_RES_BETA_PRIME
    *pc_G_TO_MPS2))), MIN_dAY_FROM_BETAPRA,
    MAX_dAY_FROM_BETAPRA);

ssYawAcc2SWARateFraction=AYC_INTPOL2(ss_mps_REF
    _VELOCITY_YC, YawAcc2SWARate_x_tab,YawAcc2SWA
    Rate_y_tab, YawAcc2SWARate_len);

ssBetaDotRA2SWARateFraction=AYC_INTPOL2(ss_mps_REF
    _VELOCITY_YC, BetaDotRA2SWARate_x_tab,
    BetaDotRA2SWARate_y_tab, BetaDotRA2SWARate
    len);

```

```

ssDayFrmYwAccScldToSWA=ssDayFromYawAcc*ssYawAcc2
    SWARateFraction/(1024.0);

ssDayFrmBetaPScldToSWA=ssDayFromBetaPRA*ssBetaDot
    RA2SWARateFraction/(1024.0);

ssFinalMaxDayDt=max(dAYFromSWA, ssDayFrmYwAccScld
    ToSWA);

ssFinalMaxDayDt=max(ssFinalMaxDayDt, ssDayFrmBeta
    PScldToSWA);

ssFinalMaxRtLmtdDayDt=max(POSRateLimiteddAYFromSWA,
    ssDayFrmYwAccScldToSWA); ssFinalMaxRtLmtdDay
    Dt=max(ssFinalMaxRtLmtdDayDt,ssDayFrmBetaPScld
    ToSWA);

ssFinalMinDayDt=min(dAYFromSWA, ssDayFrmYwAccScld
    ToSWA);

ssFinalMinDayDt=min(ssFinalMinDayDt, ssDayFrmBeta
    PScldToSWA);

/*Since NEGRateLimiteddAYFromSWA is limited to
positive values, use MAX of negated DAY's derived
from YawAcc and BetaP*/

ssFinalMinRtLmtdDayDt=max(NEGRateLimiteddAYFromSWA,
    ((-1)*ssDayFrmYwAccScldToSWA));

ssFinalMinRtLmtdDayDt=max(ssFinalMinRtLmtdDayDt,
    ((-1)*ssDayFrmBetaPScldToSWA));

```

**[0079]** In step 196 a rough estimate of the mu level of the road surface is determined. The primary objective is not precise estimation of the mu level (although one could be used), but an indication of the lateral forces expected to be experienced after the transition. The Rough MU estimate is allowed to increase with the magnitude of the lateral acceleration. The value is held high for the period ROUGH\_MU\_HOLD\_TIME following any loop in which the Rough MU estimate is increased by the lateral acceleration. Then it is

allowed to ramp down at a predefined ramp rate. The important property for this calculation is that it preserves for the duration of the event, the lateral forces experienced immediately before the transition. This information is then used as a predictor of what the forces are likely to be seen after the transition. This is in turn used to determine the pressure build profile during the transition. The Rough MU signals are bounded between 1.0 and 0.1.

**[0080]** A Rough MU Upper is determined by:

```

if (FLT_LAT_ACC > RoughMUUpper*G_TO_MPS2)
{
    ROUGH_MU_UPPER_TIMER=0;
    RoughMUUpper=FLT_LAT_ACC/G_TO_MPS2;
}
else
{
    if (ROUGH_MU_UPPER_TIMER<ROUGH_MU_HOLD_TIME)
        ROUGH_MU_UPPER_TIMER++;
    else ss_RoughMUUpper--=(ROUGH_MU_RAMP_DOWN
        _RATE*LOOP_TIME);
}

if (RoughMUUpper>max_RoughMU) RoughMUUpper=max
_RoughMU;
if (RoughMUUpper<min_RoughMU) RoughMUUpper
=min_RoughMU;

```

**[0081]** A Rough MU Lower is determined by:

```

if (-FLT_LAT_ACC>RoughMULower*G_TO_MPS2)
{
    ROUGH_MU_LOWER_TIMER=0;
    RoughMULower=-FLT_LAT_ACC/G_TO_MPS2;
}
else
{
    if (ROUGH_MU_LOWER_TIMER<ROUGH_MU_HOLD_TIME)
        ROUGH_MU_LOWER_TIMER++;
    else ss_RoughMULower--=(ROUGH_MU_RAMP_DOWN
        _RATE*LOOP_TIME);
}

```

```

    }
    if (RoughMULower > max_RoughMU) RoughMULower = max_Rough
    MU;
    if (ss_RoughMULower < min_RoughMU) RoughMULower = min
    _RoughMU;

```

**[0082]** A Final Rough MU is determined by:

```
RoughMU = max(RoughMUUpper, RoughMULower);
```

**[0083]** In step 198 a Matching Pressure is determined. The matching pressure is calculated based on the Rough MU level. Higher mu levels result in higher matching pressure. The matching pressure is used in the prediction strategy to compensate for the finite build rate available from the hydraulic system. The matching pressure is determined in the following:

```

MatchingPressure = BASE_MATCHING_PRESSURE + RoughMU *
    MATCHING_PRESSURE_MU_GAIN;
if (MatchingPressure < 0.0) MatchingPressure = 0.0;

```

**[0084]** In step 200 a predicted SWA sign change is determined. In this section two flags are calculated POS\_SWA\_EXPECTED and NEG\_SWA\_EXPECTED. POS\_SWA\_EXPECTED is set to true if the steering wheel angle is positive in the current loop or expected to be positive within a certain period of time. This period of time or signal lead is mu dependent and is intended to allow more lead in building the caliper precharge pressure for higher mu estimates. It has been experimentally determined that has indicated that on high mu and for large steering rates, initiating the pressure build after the sign crossing of the steering wheel angle may not allow adequate pressure to be built during the transition to cancel the resulting dynamic oversteer. A similar

calculation is performed for the negative direction.  
The prediction of sign change of SWA is determined by:

```
POS_SWA_EXPECTED=(RoughMU*FILTERED_STEER_ANGLE_RATE
*EXPECTED_SWA_LEAD_TIME_MU_GAIN+min(STEERING
WHEEL_ANGLE, MAX_THRESHOLD_FOR_PREDICTED_POS
_SWA)>MIN_POS_SWA_THRESHOLD)|| (STEERING_WHEEL
_ANGLE>MIN_POS_SWA_THRESHOLD);
```

```
NEG_SWA_EXPECTED=(RoughMU*FILTERED_STEER_ANGLE_RATE
*EXPECTED_SWA_LEAD_TIME_MU_GAIN+max(STEERING
WHEEL_ANGLE, MIN_THRESHOLD_FOR_PREDICTED_POS
_SWA)<MAX_POS_SWA_THRESHOLD)|| (STEERING_WHEEL
_ANGLE<MAX_POS_SWA_THRESHOLD);
```

**[0085]** In step 202 the RSC Precharge Status flags are enabled. The enable flag is true in any loop for which the precharge pressure can be implemented. In the present example the vehicle should be above a minimum speed and the rollover stability control (RSC) system must not be disabled for the flag to be true, otherwise it will be forced to a false state. A recent PID event or a combination of sufficiently high steering 100 velocities and recent large lateral acceleration may also be required for the flag to be set true.

**[0086]** Several steps are used in setting the flags. First, it is determined whether a rapid change in lateral acceleration has recently occurred. This is used as an indicator that a transitional maneuver is occurring. The code for such is:

```
if(max(FinalMaxDAyDt,-FinalMinDAyDt)>SpeedDpndntAy
Chng)Recent_Rapid_AY_Counter=RECENT_RAPID_AY
_TIME;
else if (Recent_Rapid_AY_Counter>0)Recent_Rapid_AY
_Counter--;
RECENT_RAPID_AY=Recent_Rapid_AY_Counter > 0;
```

[0087] Next it is determined whether a large rough mu has recently been experienced. This flag is used to indicate that a near limit condition is or has recently occurred.

[0088] RECENT\_LARGE\_ROUGH\_MU=RoughMU>LARGE\_ROUGH\_MU;

[0089] Then it is determined whether a moderate linear slip is or has recently occurred at the front axle as in:

```
if (ABS(FrontLnrSlipAngle)>((FRNT_SLIP_ANGLE
    _THRESHOLD_POS_SIG_DYN*FT_CRNRG_CMPLNCE)))
    RECENT_MODERATE_FRONT_AXLE_LNR_SLIP_COUNTER=
    (RECENT_MODERATE_FRONT_LINEAR_SLIP/LOOP_TIME);
else
    if
    (RECENT_MODERATE_FRONT_AXLE_LNR_SLIP_COUNTER>0)
        RECENT_MODERATE_FRONT_AXLE_LNR_SLIP_COUNTER--;
```

[0090] Then, it is determined whether a recent absolutely lifted has been determined for a wheel OR whether a recent pre lift has been sensed. This is set forth in code by:

```
if (((FINAL_WHEEL_LIFT_STATUS[FL]==ABSOLUTELY
    _LIFTED)&&(FLT_LAT_ACC>S16_RND(0.85/g_RES_LAT
    _ACC)))||WDA_LIFT_SUSPECTED[FL])
{
    RECENT_LEFT_WHEEL_LIFT_TIMER=RECENT_WHEEL_LIFT
    _TIME;
}
else if (RECENT_LEFT_WHEEL_LIFT_TIMER)RECENT_LEFT
    _WHEEL_LIFT_TIMER--;

if (RECENT_LEFT_WHEEL_LIFT_TIMER)RECENT_LEFT_WHEEL
    _LIFT=TRUE;
else
    RECENT_LEFT_WHEEL_LIFT=FALSE;

if
    (((FINAL_WHEEL_LIFT_STATUS[FR]==ABSOLUTELY_LIFTED)
```



```

        &&(FLT_LAT_ACC<S16_RND((-0.85)/g_RES_LAT
        _ACC)))||WDA_LIFT_SUSPECTED[FR])
    {
        RIGHT_WHEEL_LIFT_TIMER=RECENT_WHEEL_LIFT
        _TIME;
    }
    else
        (RECENT_RIGHT_WHEEL_LIFT_TIMER)RECENT_RIGHT_WHEEL
        _LIFT_TIMER--;

    if (RECENT_RIGHT_WHEEL_LIFT_TIMER)RECENT_RIGHT
        _WHEEL_LIFT=TRUE;
    else
        RECENT_RIGHT_WHEEL_LIFT=FALSE;

```

**[0091]** Another factor is if significant dynamics are present which might warrant a precharge intervention. The significant dynamics variable is an intermediate variable taking into consideration the above recent rapid lateral acceleration, large rough mu, linear slip, lift status, and prelift sensing. The code is implemented as:

```

SIGNIFICANT_DYNAMICS=
    (RECENT_PID_CTRL
    ||(RECENT_RAPID_AY&&RECENT_LARGE_ROUGH_MU)
    ||((max(FinalMaxDAyDt,-FinalMinDAyDt)>Speed
        DpndntAyChng)&&(RECENT_MODERATE_FRONT
        AXLE_LNR_SLIP_COUNTER>0))
    ||(FL_Precharge_Press)
    ||(FR_Precharge_Press)
    ||RECENT_LEFT_WHEEL_LIFT
    ||RECENT_RIGHT_WHEEL_LIFT
    ||Predicted_AY_from_SWA>=1.4*g&&FrontLnrSlip
        Angle<=-0.85*FT_CRNRG_CMPLNCE&&WHEEL
        DEPARTURE_ANGLE>=0.1
    ||Predicted_AY_from_SWA<=-1.4*g&&FrontLnrSlip
        Angle>=0.85*FT_CRNRG_CMPLNCE&&WHEEL
        _DEPARTURE_ANGLE<=-0.1
    )

```

**[0092]** Then based on the significant dynamics variable, the precharge pressures and other variables by the following:

```

if (SIGNIFICANT_DYNAMICS
    &&(REF_VELOCITY_YC>VELOCITY_MIN_10
    ||((FL_Precharge_Press>0||FR_Precharge
        Press>0)
    &&(REF_VELOCITY_YC>VELOCITY_MIN_TO_EXIT
        _7)))
    && !Get_rsc_disabled
    && !Get_ayc_reverse_movement
    && !Status_first_run
    && !RSC_Disable_after_sensor_self_test
)
    ENABLE_RSC_PRECHARGE=TRUE;
else
    ENABLE_RSC_PRECHARGE=FALSE;

```

**[0093]** In step 204 Precharge Timers are managed. Precharge timers are used to allow the precharge strategy to occur during or immediately after the transitional portion of an event. This serves as additional screening criteria to prevent unnecessary precharge interventions.

**[0094]** If the rear linear slip angle is consistent with a left hand turn or the significant dynamics flag indicates there is no significant dynamic activity, the left timer is zeroed out. Otherwise it is incremented to the maximum precharge time. If the rear linear slip angle is consistent with a right hand turn or the significant dynamics flag indicates there is no significant dynamic activity, the right timer is zeroed out. Otherwise it is incremented to the maximum precharge time. This ensures that within a defined period of time, after a dynamic event, a precharge event will not be initiated, reducing the possibility of unnecessary intervention. This is set forth as:

```

If (RearLnrSlipAngle<=0
    ||!SIGNIFICANT_DYNAMICS
    ||RECENT_PID_ACTIVATION
    ||RECENT_RIGHT_WHEEL_LIFT)

```

```

        FL_Precharge_Timer=0;
    else if (FL_Precharge_Timer<MAX_PRECHARGE_TIME)
        FL_Precharge_Timer++;

    if (RearLnrSlipAngle>=0
        || !SIGNIFICANT_DYNAMICS
        || RECENT_PID_ACTIVATION
        || RECENT_LEFT_WHEEL_LIFT)
        FR_Precharge_Timer=0;
    else if (FR_Precharge_Timer<MAX_PRECHARGE_TIME)
        FR_Precharge_Timer++;

```

**[0095]** In step 206 individual caliper pre-charge status flags are determined. The status flags indicate whether the pre-charge function is active on a given wheel in a given loop. Pre-charge must be enabled in any loop for which the pre-charge strategy is active. If the precharge timer on a given wheel is below the maximum precharge time, the function can become activated on that wheel based on steering information or side slip angle information. Additionally, the pre-charge function will remain active on a specific wheel as long as there is a non-zero pressure being requested on that wheel.

**[0096]** The continuous time of consistent sign for SSLIN Front or Rear is monitored.

```

    if (FrontLnrSlipAngle>MIN_SLIP_ANGLE_FOR_PRCHG_ACT
        || RearLnrSlipAngle>MIN_SLIP_ANGLE_FOR
            _PRCHG_ACT)
    {
        if (PSTV_SS_LIN<MIN_CONT_TIME_WITH_CONSISTENT_SIGN)
            PSTV_SS_LIN++;
    }
    else if (PSTV_SS_LIN>0) PSTV_SS_LIN--;

    if (FrontLnrSlipAngle<MIN_SLIP_ANGLE_FOR_PRCHG
        _ACT
        || RearLnrSlipAngle<MIN_SLIP_ANGLE_FOR
            _PRCHG_ACT)

```

```

{
if (NGTV_SS_LIN<MIN_CONT_TIME_WITH_CONSISTENT_SIGN)
    NGTV_SS_LIN++;
}
else if (NGTV_SS_LIN>0) NGTV_SS_LIN--;

```

**[0097]** Then the front left precharge active status flags are determined.

```

FL_PRECHARGE_ACTIVE=ENABLE_RSC_PRECHARGE&&(((NEG
    _SWA_EXPECTED
/*Allow FL transition control to activate if
FR has transition pressure and SWA direction
is heading towards the right (heading towards
a negative value)*/
    &&((FinalMinDayDt<(-SpeedDpndntAyChng))
    ||(FR_Precharge_Press>(0.0) )))
    ||(RearLnrSlipAngle>MIN_SLIP_ANGLE
        _FOR_PRCHG_ACT)
    ||(FrontLnrSlipAngle>MIN_SLIP_ANGLE_FOR
        _PRCHG_ACT))
    &&FL_Precharge_Timer<MAX_PRECHARGE
        _TIME)
    ||FL_Precharge_Press>0.0)
    ||PSTV_SS_LIN>=MIN_CONT_TIME_WITH
        _CONSISTENT_SIGN

```

**[0098]** When the front right precharge active status flags are determined:

```

FR_PRECHARGE_ACTIVE=ENABLE_RSC_PRECHARGE&&(((POS
    _SWA_EXPECTED
/*Allow FR transition control to activate if FL has
transition pressure and SWA direction is heading
towards the left (heading towards a positive
value)*/
    &&((FinalMaxDayDt>SpeedDpndntAyChng)|| (FL
        _P_recharge_Press>(0.0)))
    ||(RearLnrSlipAngle<MIN_SLIP_ANGLE_FOR_PRCHG
        _ACT)
    ||(FrontLnrSlipAngle<MIN_SLIP_ANGLE_FOR_PRCHG
        _ACT))&&FR_Precharge_Timer<MAX_PRECHARGE
        _TIME)
    ||FR_Precharge_Press>0.0);

```

||PSTV\_SS\_LIN>=MIN\_CONT\_TIME\_WITH\_CONSISTENT  
\_SIGN

**[0099]** In step 208 calculate moderate lateral acceleration transition flags. If a lateral acceleration of a defined magnitude precedes a transition controller event associated with the opposite direction of turning within a predefined period of time, then a moderate transition flag is set for that direction until the transition controller event has been exited. These flags are used in the wheel departure angle calculation to bias any changes in total roll angle and reference bank angle towards an increasing roll signal for control.

**[00100]** First a counter (moderate-positive-AY-counter) is implemented which is set to a transition hold time any time the filtered lateral acceleration exceeds the transition threshold, and is decremented to zero by one count per if the lateral acceleration is not exceeding the transition lateral acceleration threshold.

```
if (FLT_LAT_ACC > TRANSITION_AY_THRESHOLD) MODERATE  
_POSITIVE_AY_COUNTER = TRANSITION_HOLD_TIME;  
else if (MODERATE_POSITIVE_AY_COUNTER > 0) MODERATE  
_POSITIVE_AY_COUNTER--;
```

**[00101]** Next, the recent positive lateral acceleration flag is calculated. This flag is set to true if the counter is greater than zero, otherwise it is set to false for a zero value of the counter.

**[00102]**

```

if(MODERATE_POSITIVE_AY_COUNTER>0)RECENT_POSITIVE
  _AY=TRUE;
else RECENT_POSITIVE_AY=FALSE;

```

**[00103]** The counter is now implemented for moderate negative lateral acceleration values. The negative counter is set to a transition hold time any time the filtered lateral acceleration falls below the negative of the transition threshold, and is decremented to zero by one count per if the lateral acceleration is not below the negative of the transition threshold.

```

if(FLT_LAT_ACC<-TRANSITION_AY_THRESHOLD)
  MODERATE_NEGATIVE_AY_COUNTER=TRANSITION
  _HOLD_TIME;
else if(MODERATE_NEGATIVE_AY_COUNTER>0)MODERATE
  _NEGATIVE_AY_COUNTER--;

```

**[00104]** Similar to above, the recent negative lateral acceleration flag is calculated. This flag is set to true if the counter is greater than zero, otherwise it is set to false for a zero value of the counter.

```

if(MODERATE_NEGATIVE_AY_COUNTER>0)RECENT_POSITIVE
  _AY=TRUE;
else RECENT_POSITIVE_AY=FALSE;

```

**[00105]** Next, the right to left transition flag is determined based on the recent negative lateral acceleration flag and the front right transition controller active flag. If both flags are set to true, then the right to left transition flag is set to true. If the front right transition controller active flag is false, the right to left transition controller flag is set to false. Otherwise, the value is held from the previous loop. This results in a flag that remains true until the front right transition controller event is exited.

```

if(RECENT_NEGATIVE_AY&&FR_Precharge_Press)RIGHT_TO
    _LEFT_TRANSITION=TRUE;
else
if(!FR_Precharge_Press)RIGHT_TO_LEFT_TRANSITION
    =FALSE;
else
RIGHT_TO_LEFT_TRANSITION=RIGHT_TO_LEFT_TRANSITION;

```

**[00106]** Now the same calculation is performed for a left to right transition flag. If the recent negative lateral acceleration flag is true and the front left transition controller active flag is true, then the left to right transition flag is set to true. If the front left transition controller active flag is false, the left to right transition controller flag is set to false. Otherwise, the value is held from the previous loop. This results in a flag that remains true until the front left transition controller event is exited.

```

if(RECENT_POSITIVE_AY&&FL_Precharge_Press) LEFT_TO
    _RIGHT_TRANSITION=TRUE;
else if(!FL_Precharge_Press) LEFT_TO_RIGHT
    _TRANSITION=FALSE;
else LEFT_TO_RIGHT_TRANSITION=LEFT_TO_RIGHT
    _TRANSITION;

```

**[00107]** Then a timer is created to limit duration of high target slip request in the following:

```

if
    (union_PRECHARGE_FLAGS.st_PRECHARGE.bf_bool_RIGHT
        _TO_LEFT_TRANSITION==TRUE)
{
    if (R2L_TRANSITION_COUNTER<TIME_LIMIT_TO_REQ_EXCSV
        _SLIP_AFTER_TRANSITION)R2L_TRANSITION_COUNTER
        ++;
}

else

{
    R2L_TRANSITION_COUNTER=0;
}

```

```

if
  (union_PRECHARGE_FLAGS.st_PRECHARGE.bf_bool_LEFT_TO
   _RIGHT_TRANSITION==TRUE)
{
  if (L2R_TRANSITION_COUNTER<TIME_LIMIT_TO_REQ
      _EXCSV_SLIP_AFTER_TRANSITION)L2R
      _TRANSITION_COUNTER++;
}
else
{
  L2R_TRANSITION_COUNTER=0;
}

```

**[00108]** In step 210 the rate of change of relative roll angle is determined. The relative roll velocity is calculated by simple differentiation of the relative roll angle placed in series with a first order low pass filter as in:

```

REL_ROLL_VELOCITY+=((REL_ROLL_ANGLE-Z1_REL_ROLL
  _ANGLE)/LOOP_TIME)-REL_ROLL_VELOCITY)/Rel_Roll
  _Vel_FLT_Coeff;

```

**[00109]** In step 212, the rate of change of linear front slip angle is determined. Similarly, the linear front tire side slip angle velocity is calculated by simple differentiation of the linear front side slip angle placed in series with a first order low pass filter.

```

Temp=MAX(min_Ref_Velocity_YC,Ref_Velocity_YC);
Lin_Front_Vel+=((CurrentFrontLnrLatVel-FRONT_LNR
  _LAT_VEL)*(rad_to_deg/LOOP_TIME)/Temp
  -Lin_Front_Vel)/lin_frnt_vel_filter
  _coeff;

```

**[00110]** In step 214 a lower target build rate is used, yielding larger prediction pressures when steady state limit driving is detected. This is set forth in:



```

if
((Predicted_AY_from_SWA>=1.4*g&&(FrontLnrSlipAngle
<=-0.85*FT_CRNRG_CMPLNCE))
&&(WHEEL_DEPARTURE_ANGLE>=(0.1/deg_RES_ROLL
_ANGLE)))||(Predicted_AY_from_SWA<=-.4*g
&&(FrontLnrSlipAngle>=(0.85*FT_CRNRG
_CMPLNCE)
&&(WHEEL_DEPARTURE_ANGLE<=(-0.1/deg
_RES_ROLL_ANGLE)))
)
{
    RECENT_STEADY_STATE_CORNERING_TIMER
    =RECENT_STDY_STATE_TURN_NEAR_LIMIT;
}
else
{
    if (RECENT_STEADY_STATE_CORNERING_TIMER>0)RECENT
    _STEADY_STATE_CORNERING_TIMER-
    }
    if (RECENT_STEADY_STATE_CORNERING_TIMER>0)
    {
        TrgtBldRateStdyState=TARGET_BUILD_RATE;
    } else
    {
        TrgtBldRateStdyState=Rsc_target_build
        _rate;
    }
}

```

**[00111]** In step 216 an instantaneous caliper pre-charge requested pressure levels based on relative roll information is determined. First, a pressure is calculated based on proportional plus derivative control on roll information. Relative roll angle is used for the proportional term, roll velocity is used for the derivative term. The relative roll angle is used because it provides the required level of accuracy during the critical portion of the transition, during the zero crossing, and because it is not influenced by banks and is not influenced by integrator error or errors in reference road bank. This accuracy allows tighter thresholds than would be allowable for the PID controller. Additionally, this signal offers a high

level of consistency and repeatability in the timing of the pressure build that is a critical property for the initial portion of the control intervention. The PD pressure is filtered as required to achieve an adequately smooth signal.

**[00112]** Next the derivative of the PID pressure is calculated for determining the instantaneous relative roll pressure. The instantaneous pressure provides a predictive functionality for the PD controller. It is intended to provide a mechanism to compensate for the finite rate at which pressure can be built. Additionally, it can provide a significant smoothing effect on the leading edge of the pressure trace without delaying the pressure build. This is helpful in building the required level of pressure while minimizing the excitation of the pitching mode of the vehicle. The instantaneous pressure is that pressure which would need to be in the caliper in the current loop such that if the pressure is increased at the target build rate, the matching pressure would be achieved at the same time the PID controller is expected to request the matching pressure. The point in time for which the PID controller is expected to request the matching pressure is obtained by taking the derivative of the PID pressure and projecting forward to the point in time when the intersects the matching pressure.

**[00113]** Front left relative roll instantaneous pressure is determined by:

$$\begin{aligned} \text{FLRelRollPDPress} = & (\text{KP\_REL\_ROLL} * (-\text{REL\_ROLL\_ANGLE} \\ & - \text{REL\_ROLL\_DB}) - \text{KD\_REL\_ROLL} * \text{FLT\_ROLL\_RATE} \\ & - \text{FLRelRollPDPressZ1}) * \text{REL\_ROLL} \end{aligned}$$

\_PD\_PRESS\_FILT

```
FLRelRollPDPress=min(FLRelRollPDPress, MAX_FL_REL  
_ROLL_PD_PRESS);
```

```
FLRelRollPDPress=max(FLRelRollPDPress, MIN_FL_REL  
_ROLL_PD_PRESS);
```

```
dFLRelRollPDPress=FLRelRollPDPress-FLRelRollPD  
PressZ1;
```

```
FLRelRollPDPressZ1=FLRelRollPDPress;
```

```
FLRelRollInstPress=MatchingPressure-(Matching  
Pressure-FLRelRollPDPress)*TARGET_BUILD  
_RATE /max(TARGET_BUILD_RATE, dFLRelRoll  
PDPress /LOOP_TIME);
```

**[00114]** Then the Front Right relative roll instantaneous pressure is determined by:

```
FRRelRollPDPress+=(KP_REL_ROLL*(REL_ROLL_ANGLE-REL  
_ROLL_DB)+KD_REL_ROLL*FLT_ROLL_RATE-FRRelRoll  
PDPressZ1)*REL_ROLL_PD_PRESS_FILT)
```

```
FRRelRollPDPress=min(FRRelRollPDPress, MAX_FL_REL  
_ROLL_PD_PRESS);
```

```
FRRelRollPDPress=max(FRRelRollPDPress, MIN_FL_REL  
_ROLL_PD_PRESS);
```

```
dFRRelRollPDPress=FRRelRollPDPress-FRRelRollPD  
PressZ1;
```

```
FRRelRollPDPressZ1=FRRelRollPDPress;
```

```
FRRelRollInstPress=MatchingPressure-(Matching  
Pressure-FRRelRollPDPress)*TARGET_BUILD_RATE  
/max(TARGET_BUILD_RATE, dFRRelRollPDPress  
/LOOP_TIME);
```

**[00115]** In step 218 the conditions for applying the Yaw Rate PID controller to prevent a pressure increase from the yaw rate controller when driver is steering out of a turn are determined.

**[00116]** When transitioning from a right to a left turn, determine when change in SWA direction to the left has been well established; use as a condition for blocking the YAW RATE PD controller from further increasing transition pressure on FL wheel

```

if ((STEERING_WHEEL_ANGLE-Z1_STEERING_WHEEL_ANGLE)
    >=0)
{
    if (INCREASING_SWA_COUNTER<(UPPER_BOUND_ON
        _TIME_FOR_DETECTION/LOOP_TIME_SEC))
    {
        INCREASING_SWA_COUNTER++;
    }
    if (POSITIVE_DELTA_SWA_INT<UPPER_BOUND_ON
        _DELTA_SWA_INT)
    {
        //Accumulate positive delta SWA POSITIVE
        _DELTA_SWA_INT+=(STEERING_WHEEL_ANGLE-Z1
            _STEERING_WHEEL_ANGLE);
    }
}
else // SWA is heading to the right
{
    INCREASING_SWA_COUNTER=0; // Reset increasing
    SWA counter POSITIVE_DELTA_SWA_INT=0; // Reset
    delta SWA integration
}
if
(LARGE_LAT_ACC_COUNTER==0||(Predicted_AY_from_SWA>-
    AY_FROM_SWA_THRESH_FOR_DIRECTION_CHANGE
    &&INCREASING_SWA_COUNTER>=TIME_FOR_DETECTING
    _SWA_DIRECTION_CHANGE&&POSITIVE_DELTA_SWA_INT
    >=(SWA_VEL_FOR_DETECTING_SWA_DIRECTION_CHANGE
    *TIME_FOR_DETECTING_SWA_DIRECTION_CHANGE))
)
// Disable YawRate PD controller on FL from
increasing transition pressure if LatAcc at CG
magnitude did not exceed lateral accel threshold
(tunable parameter set to 0.5 g's) within the past
3 seconds or anytime the predicted Ay from SWA is
greater than (-0.5 g) and SWA and SWA_VEL indicate
a consistent positive sign trend
ROLL_FLAG.bf_bool_FL_DISABLE_YR_PD_PR_INC=TRUE;
}
else
{
    ROLL_FLAG.bf_bool_FL_DISABLE_YR_PD_PR_INC=FALSE;
}

```

}  
**[00117]** In the case when transitioning from a left to a right turn, it is determined when a change in SWA direction to the right has been well established; to use as a condition for blocking the YAW RATE PD controller from further increasing this transition pressure on FR wheel, the following is used:

```

if (STEERING_WHEEL_ANGLE-Z1_STEERING_WHEEL_ANGLE
    <=0)
{
    if
    (DECREASING_SWA_COUNTER<UPPER_BOUND_ON_TIME_FOR
      DETECTION/LOOP_TIME_SEC)
    {
        // Increase counter each loop delta
        SWA is not negative DECREASING_SWA
        _COUNTER ++;
    }
    if (NEGATIVE_DELTA_SWA_INT>-UPPER_BOUND_ON
        DELTA_SWA_INT)
    {
        // Accumulate positive delta SWANEUTRAL
        DELTA_SWA_INT+=STEERING_WHEEL_ANGLE-Z1
        STEERING_WHEEL_ANGLE;
    }
}
else // SWA is heading to the left
DECREASING_SWA_COUNTER=0; // Reset decreasing
SWA counter
NEGATIVE_DELTA_SWA_INT=0; // Reset delta SWA
integration
}

if
(LARGE_LAT_ACC_COUNTER==0||(Predicted_AY_from_SWA
  <AY_FROM_SWA_THRESH_FOR_DIRECTION_CHANGE&&
  DECREASING_SWA_COUNTER>=TIME_FOR_DETECTING_SWA
  DIRECTION_CHANGE&&NEGATIVE_DELTA_SWA_INT<=
  (SWA_VEL_FOR_DETECTING_SWA_DIRECTION_CHANGE
  *TIME_FOR_DETECTING_SWA_DIRECTION_CHANGE))
)
//Disable YawRate PID controller on FR from
increasing transition pressure if LatAcc at CG
magnitude did not exceed lateral accel threshold

```

(tunable parameter set to 0.5 g's) within the past 3 seconds or anytime the predicted Ay from SWA is less than +0.5 g and SWA and SWA\_VEL indicate a consistent negative sign trend

```
{
    ROLL_FLAG.bf_bool_FR_DISABLE_YR_PD_PR_INC
        =TRUE;
}
else
{
    ROLL_FLAG.bf_bool_FR_DISABLE_YR_PD_PR_INC
        =FALSE;
}
```

**[00118]** Then the instantaneous caliper precharge requested pressure levels based on yaw rate are determined.

```
FLYawRatePDPress=LIMIT((FL_YAW_RATE_PD_PRESS_Z1+
    (Rsc_kp_yaw_rate*(-FLT_YAW_RATE-SpeedDpndnt
    YRDB))-Rsc_kd_yaw_rate*SLOW_FLT_YAW_ACC-FL
    _YAW_RATE_PD_PRESS_Z1)/(1.0/YAW_RATE_PD_PRESS
    _FILT)), MIN_PD_PRESS, MAX_PD_PRESS);

dFLYawRatePDPress=FLYawRatePDPress-FL_YAW_RATE_PD
    _PRESS_Z1;

FL_YAW_RATE_PD_PRESS_Z1=FLYawRatePDPress;

#if (USE_LOWER_TARGET_BUILD_RATE_FOR_STEADY_STATE
    _TURNS) FLYawRateInstPress=MatchingPressure
    -(MatchingPressure-FLYawRatePDPress)*TrgtBld
    RateStdyState/max(TrgtBldRateStdyState, dFLYaw
    RatePDPress/LOOP_TIME_SEC));

#else
FLYawRateInstPress=MatchingPressure-(Matching
    Pressure-FLYawRatePDPress)*Rsc_target
    _build_rate/max(Rsc_target_build_rate,
    dFLYawRatePDPress/LOOP_TIME_SEC));

#endif
```

```
FRYawRatePDPress=LIMIT((FR_YAW_RATE_PD_PRESS
    _Z1+(Rsc_kp_yaw_rate*(FLT_YAW_RATE
    -SpeedDpndntYRDB)+Rsc_kd_yaw_rate*SLOW
    _FLT_YAW_ACC)-FR_YAW_RATE_PD_PRESS_Z1)/
    (1.0/YAW_RATE_PD_PRESS_FILT)), MIN_PD
```

```

        _PRESS, MAX_PD_PRESS);

dFRYawRatePDPress=FRYawRatePDPress-FR_YAW_RATE
_PD_PRESS_Z1;

FR_YAW_RATE_PD_PRESS_Z1=FRYawRatePDPress;

#if (USE_LOWER_TARGET_BUILD_RATE_FOR_STEADY
_STATE_TURNS) FRYawRateInstPress=Matching
Pressure-(MatchingPressure-FRYawRatePD
Press)*ss_bps_TrgtBldRateStdyState/max
(TrgtBldRateStdyState, dFRYawRatePDPress*
(1.0/LOOP_TIME_SEC));

#else

FRYawRateInstPress=MatchingPressure-(Matching
Pressure-FRYawRatePDPress)*Rsc_target
_build_rate/max(Rsc_target_build_rate,
dFRYawRatePDPress*(1.0/LOOP_TIME_SEC));

#endif

```

**[00119]** In step 220 the instantaneous caliper precharge pressures based on front linear slip angle are determined. In this section, calculations are performed which are similar to those used for the Relative Roll PD pressure. Instead of Relative Roll angle for the proportional term, the linear side slip angle of the front axle is used. Instead of roll velocity, the rate of change of the front linear slip angle is used for the derivative term.

**[00120]** First, the Front Left front "front linear slip angle" instantaneous pressure is determined by:

```

FLSSLinFrontPDPress+=(KP_SS_LIN_FRONT*(FrontLnrSlip
Angle-
SS_LIN_FRONT_DB)+KD_SS_LIN_FRONT*Lin_Front_Vel-FLSS
LinFrontPDPressZ1)*SS_LIN_PD_PRESS_FILT;

dFLSSLinFrontPDPress=FLSSLinFrontPDPress-FLSS
LinFrontPDPressZ1;

```

FLSSLinFrontPDPressZ1=FLSSLinFrontPDPress;

FLSSLinFrontInstPress=MatchingPressure-(Matching  
Pressure-FLSSLinFrontPDPress)\*TARGET\_BUILD  
\_RATE/max(TARGET\_BUILD\_RATE, dFLSSLinFront  
PDPress /LOOP\_TIME);

**[00121]** Then Front Right front "front linear slip  
angle" instantaneous pressure is determined by:

FRSSLinFrontPDPress+=(KP\_SS\_LIN\_FRONT\*(-Front  
LnrSlipAngle-SS\_LIN\_FRONT\_DB)-KD\_SS\_LIN\_FRONT\*  
Lin\_Front\_Vel-FRSSLinFrontPDPressZ1)\*SS\_LIN  
\_PD\_PRESS\_FILT;

dFRSSLinFrontPDPress=FRSSLinFrontPDPress-FRSSLin  
FrontPDPressZ1;

FRSSLinFrontPDPressZ1=FRSSLinFrontPDPress;

FRSSLinFrontInstPress=MatchingPressure-(Matching  
Pressure-FRSSLinFrontPDPress)\*TARGET\_BUILD  
\_RATE/max(TARGET\_BUILD\_RATE, dFRSSLinFront  
PDPress/LOOP\_TIME);

**[00122]** In step 222 the instantaneous build rate based  
on the slope of instantaneous PID pressure in relation  
to the matching pressure and the length of time it would  
take for the prior value of precharge pressure to reach  
the matching pressure level is determined.

**[00123]** First the front left FL instantaneous build  
rate based on RelRoll PD information is determined.

PDRelRollMatchPressDelta[FL]=MatchingPressure-  
FLRelRollPDPress;  
if (PDRelRollMatchPressDelta[FL]<=0)  
{  
InstRelRollBuildRate[FL]=(MIN\_INST\_BUILD\_RATE);  
}  
else  
{  
//Put lower limit on denominator to avoid  
divide by zero  
PDRelRollMatchPressDelta[FL]=max(PDRelRoll



```

        MatchPressDelta[FL], (MIN_MATCH_PRESS_2
        _PD_PRESS_DELTA));
//Calculate RelRoll based build rate, and
bound between 100 and 300 bar/s
InstRelRollBuildRate[FL]=
    LIMIT((((MatchingPressure-FL_Precharge
    _Press)*(dFLRelRollPDPress*(1.0/LOOP
    TIME)))/PDRelRollMatchPressDelta
    [FL]), (MIN_INST_BUILD_RATE), (MAX
    _INST_BUILD_RATE));
    }

```

**[00124]** Then the FL instantaneous build rate based on Yaw Rate PD information is determined.

```

ssPDYawRateMatchPressDelta[FL]=ss_MatchingPressure
    -ss_FLYawRatePDPress;
if (ssPDYawRateMatchPressDelta[FL]<=0)
{
    ssInstYawRateBuildRate[FL]=MIN_INST_BUILD_RATE
        /bar_RES_BRAKE_PRESSR;
}
else
{
    //Put lower limit on denominator to avoid
    divide by zeros
    ssPDYawRateMatchPressDelta[FL]=max(ssPDYawRate
    MatchPressDelta[FL],
    MIN_MATCH_PRESS_2_PD_PRESS_DELTA/bar_RES_BRAKE
    _PRESSR);
    //Calculate YawRate based build rate, and
    bound between 100 and 300 bar/s
    ssInstYawRateBuildRate[FL]
        =LIMIT((((ss_MatchingPressure-ss_FL
        _Precharge_Press)*ss_dFLYawRatePDPress
        *1.0/p_LOOP_TIME_SEC/ssPDYawRateMatch
        PressDelta[FL]),
        MIN_INST_BUILD_RATE/bar_RES_BRAKE
        _PRESSR,
        MAX_INST_BUILD_RATE/bar_RES_BRAKE
        _PRESSR);
}

```

**[00125]** Then the FL instantaneous build rate based on SSLinFront PD information is determined.

**[00126]** Then the FL instantaneous build rate based on SSLinFront PD information is determined.

```
PDSSLinMatchPressDelta[FL]=MatchingPressure-FLSSLinFrontPDPress;
if (PDSSLinMatchPressDelta[FL]<=0)
{
    InstSSLinFrntBuildRate[FL]=(MIN_INST_BUILD_RATE);
}
else
{
    //Put lower limit on denominator to avoid divide by zero
    PDSSLinMatchPressDelta[FL]=max(PDSSLinMatchPressDelta[FL], (MIN_MATCH_PRESS_2_PD_PRESS_DELTA));
    //Calculate SSLinFront based build rate, and bound between 100 and 300 bar/s
    InstSSLinFrntBuildRate[FL]=
        LIMIT((((MatchingPressure-FL_Precharge_Press)*(dFLSSLinFrontPD_Press*(1.0/LOOP_TIME)))) /
        PDSSLinMatchPressDelta[FL]),
        (MIN_INST_BUILD_RATE), (MAX_INST_BUILD_RATE));
}
```

**[00127]** Then a final FL instantaneous build rate by taking maximum of RelRoll and SSLinFront rates.

```
InstBuildRate[FL]=max(InstRelRollBuildRate[FL],
    InstSSLinFrntBuildRate[FL]);
```

**[00128]** The same process is repeated for the front right FR. First, the FR instantaneous build rate based on RelRoll PD information is determined.

```

PDRollMatchPressDelta[FR]=MatchingPressure-FR
  RelRollPDPress;
if (PDRollMatchPressDelta[FR]<=0)
{
    InstRelRollBuildRate[FR]=(MIN_INST_BUILD
        _RATE);
}
else
{
    //Put lower limit on denominator to avoid
    divide by zero
    PDRollMatchPressDelta[FR]=max(PDRoll
        RollMatchPressDelta[FR], (MIN_MATCH
        _PRESS_2_PD_PRESS_DELTA));

    //Calculate RelRoll based build rate, and
    bound between 100 and 300 bar/s
    InstRelRollBuildRate[FR]=
        LIMIT((((MatchingPressure-FR
        _Precharge_Press)
        (dFRRelRollPDPress*(1.0/LOOP_TIME)))
        /PDRollMatchPressDelta[FR]),
        (MIN_INST_BUILD_RATE), (MAX_INST
        _BUILD_RATE));
}

```

**[00129]** Then the calculation of FR instantaneous build rate based on Yaw Rate PD information is performed by:

```

ssPDYawRateMatchPressDelta[FR]=ss_MatchingPressure
    -ss_FRYawRatePDPress;
if (ssPDYawRateMatchPressDelta[FR]<=0)
{
    ssInstYawRateBuildRate[FR]=MIN_INST_BUILD
        _RATE/bar_RES_BRAKE_PRESSR;
}
else
{
    //Put lower limit on denominator to avoid
    divide by zero
    ssPDYawRateMatchPressDelta[FR]=max(ssPD
        YawRateMatchPressDelta[FR],
        MIN_MATCH_PRESS_2_PD_PRESS_DELTA/bar_RES_BRAKE
        _PRESSR);
    //Calculate YawRate based build rate, and
    bound between 100 and 300 bar/s
    ssInstYawRateBuildRate[FR]=

```

```

LIMIT(((ss_MatchingPressure-ss_FR
_Precharge_Press)*
ss_dFRYawRatePDPress*1.0/p_LOOP_TIME
_SEC/
ssPDYawRateMatchPressDelta[FR]),
MIN_INST_BUILD_RATE/bar_RES
_BRAKE_PRESSR,
MAX_INST_BUILD_RATE/bar_RES
_BRAKE_PRESSR);
}

```

**[00130]** After, the FR instantaneous build rate based on SSLinFront PD information is determined by:

```

PDSSLinMatchPressDelta[FR]=MatchingPressure-FRSS
_LinFrontPDPress;
if (ssPDSSLinMatchPressDelta[FR]<=0)
{
    InstSSLinFrntBuildRate[FR]=(MIN_INST
_BUILD_RATE);
}
else
{
    //Put lower limit on denominator to avoid
    divide by zero
    PDSSLinMatchPressDelta[FR]=max(PDSSLin
    MatchPressDelta[FR],
    (MIN_MATCH_PRESS_2_PD_PRESS
    _DELTA));
    //Calculate SSLinFront based build rate,
    and bound between 100 and 300 bar/s
    InstSSLinFrntBuildRate[FR]=
    LIMIT((((MatchingPressure-FR
    _Precharge_Press)
    *(dFRSSLinFrontPDPress*(1.0/LOOP
    _TIME)))
    /PDSSLinMatchPressDelta[FR]),
    (MIN_INST_BUILD_RATE),(MAX
    _INST_BUILD_RATE) );
}

```

**[00131]** Then the final FR instantaneous build rate by taking maximum of RelRoll and SSLinFront rates

```

InstBuildRate[FR]=max(InstRelRollBuildRate[FR],
    InstSSLinFrntBuildRate[FR] );

```

[00132] In step 224, the requested pressures for each caliper are determined as follows.

[00133] Front left requested pressures are determined by:

```
FLInstPressRequest=0.0;
FLInstPressRequest=max(FLInstPressRequest, FLRel
    RollInstPress);
FLInstPressRequest=max(FLInstPressRequest, FLSS
    LinFrontInstPress);
FLInstPressRequest=max(FLInstPressRequest,ss_FLYaw
    RateInstPress);
FLInstPressRequest=min(FLInstPressRequest, Matching
    Pressure);
FLInstPressRequest=max(FLInstPressRequest, FLRel
    RollPDPress);
FLInstPressRequest=max(FLInstPressRequest, FLSS
    LinFrontPDPress);
FLInstPressRequest=max(FLInstPressRequest,ss_FLYaw
    RatePDPress);
```

[00134] Front right requested pressures are determined by:

```
FRInstPressRequest=0.0;
FRInstPressRequest=max(FRInstPressRequest, FRRel
    RollInstPress);
FRInstPressRequest=max(FRInstPressRequest, FRSS
    LinFrontInstPress);
FRInstPressRequest=max(FRInstPressRequest,ss_FRYaw
    RateInstPress);
FRInstPressRequest=min(FRInstPressRequest, Matching
    Pressure);
FRInstPressRequest=max(FRInstPressRequest, FRRel
    RollPDPress);
FRInstPressRequest=max(FRInstPressRequest, FRSS
    LinFrontPDPress);
FRInstPressRequest=max(FRInstPressRequest,ss_FRYaw
    RatePDPress);
```

[00135] In step 226 the front left caliper precharge requests are updated:

```

if (FL_PRECHARGE_ACTIVE)
{

```

**[00136]** If the RelRoll or SSLin PD pressure requests an increase, the requested pressures are ramped up.

```

if ((FLRelRollPDPress>FL_Precharge_Press)|| (FLSSLin
    FrontPDPress>FL_Precharge_Press)
    ||FLYawRatePDPress>FL_Precharge_Press)
{
    FL_Precharge_Press+=LARGE_DYNAMICS_BUILD_RATE*
        LOOP_TIME;
}

```

**[00137]** If the instantaneous pressure requests an increase, the requested pressures are ramped up.

```

else if (FLInstPressRequest>ss_FL_Precharge_Press)
{
    if (RECENT_PID_ACTIVATION[FL]||RECENT_RIGHT
        _WHEEL_LIFT )
    {
        FL_Precharge_Press+=LARGE_DYNAMICS_BUILD
            _RATE*LOOP_TIME;
    }
    else
    {
        FL_Precharge_Press+=(InstBuildRate[FL]
            *LOOP_TIME);
    }
}

```

**[00138]** If the steering information suggests a transitional event, the pressure is ramped up to a low mu dependent value in the following:

```

else if (NEG_SWA_EXPECTED
    &&FinalMinRtLmtdDAYDt >SpeedDpndntAy
        Chng
    &&FL_Precharge_Press<RoughMU*BASE
        _PRESSURE_MU_GAIN)
{
    FL_Precharge_Press+=TARGET_BUILD_RATE
        *LOOP_TIME;
}

```

}

**[00139]** If a pressure increase is not requested AND no PID pressure increase is requested, the requested pressure is ramped down.

```
else if ((FL_Precharge_Timer<PRECHARGE_BUILD_TIME)
  &&((PID_STBLZ_PRES[FL]>FL_Precharge_Press)
    ||(!PID_FL_INCREASE_REQUESTED_PRESSURE))
    &&(REL_ROLL_ANGLE>-REL_ROLL_DB)
    &&FinalMinRtLmtdDAyDt<Speed
      DpndntAyChng)
{
  FL_Precharge_Press-=TARGET_DUMP_RATE*LOOP
    _TIME;
}
```

**[00140]** Any time the timer is above the build time AND no PID pressure increase is requested AND RelRollAngle is smaller than the RelRollDB, a reduction in pressure is forced.

```
if(FL_Precharge_Timer>=PRECHARGE_BUILD_TIME
  &&(PID_STBLZ_PRES[FL]>FL_Precharge_Press
    ||!PID_INCREASE_REQUESTED_PRESSURE
      [FL])
    &&REL_ROLL_ANGLE>-REL_ROLL_DB)
{
  temp=FL_Precharge_Press/(max(1,MAX
    _PRECHARGE_TIME-FL_Precharge_Timer));
  temp=min(2*TARGET_DUMP_RATE*LOOP_TIME,
    temp);
  temp=max(TARGET_DUMP_RATE*LOOP_TIME,
    temp);
  FL_Precharge_Press-=temp;
}

FL_Precharge_Press=max(0.0,FL_Precharge
  _Press);
}

else
{
  FL_Precharge_Press=0.0;
}
```

[00141] In a similar manner the front right precharge request is updated.

```
If (FR_PRECHARGE_ACTIVE)
{
```

[00142] If the RelRoll or SSLin PD pressure requests an increase, the requested pressure is ramped up.

```
if (FRRelRollPDPress>FR_Precharge_Press||FRSSLin
    FrontPDPress>FR_Precharge_Press
    ||FRYawRatePDPress>FR_Precharge
        _Press)
{
    FR_Precharge_Press+=LARGE_DYNAMICS_BUILD_RATE
        *LOOP_TIME;
}
```

[00143] If the instantaneous pressure requests an increase, the requested pressure is ramped up.

```
else if (FRInstPressRequest>FR_Precharge_Press)
{
    if (RECENT_PID_ACTIVATION[FR]||RECENT_LEFT
        _WHEEL_LIFT)
    {
        FR_Precharge_Press+=LARGE_DYNAMICS
            _BUILD_RATE*p_LOOP_TIME_SEC;
    }
    else
    {
        FR_Precharge_Press+=(InstBuildRate[FR]*LOOP
            _TIME);
    }
}
```

[00144] If the steering information suggests a transitional event, the pressure is ramped up to a low mu dependent value



```

else if (POS_SWA_EXPECTED
      &&FinalMaxRtLmtDDayDt>SpeedDpndntAyChng
      &&FR_Precharge_Press<RoughMU*BASE_PRESSURE
        _MU_GAIN)
{
  FR_Precharge_Press+=TARGET_BUILD_RATE*LOOP
    _TIME;
}

```

**[00145]** If a pressure increase is not requested AND no PID pressure increase requested, then the requested pressure is ramped down.

```

else if (FR_Precharge_Timer<PRECHARGE_BUILD_TIME
      &&(PID_STBLZ_PRES[FR]>FR_Precharge_Press
        ||!PID_INCREASE_REQUESTED_PRESSURE
          [FR])
      &&REL_ROLL_ANGLE<REL_ROLL_DB
      &&FinalMaxRtLmtDDayDt<SpeedDpndntAyChng
      {
        FR_Precharge_Press-=TARGET_DUMP
          _RATE*LOOP_TIME;
      }

```

**[00146]** Any time the timer is above the build time AND no PID pressure increase is requested AND RelRollAngle is smaller than the RelRollDB, force a reduction in pressure.

```

if ((FR_Precharge_Timer>=PRECHARGE_BUILD_TIME
    &&(PID_STBLZ_PRES[FR]>FR_Precharge_Press
      ||!PID_INCREASE_REQUESTED_PRESSURE[FR])
    &&REL_ROLL_ANGLE<REL_ROLL_DB)
{
  temp=FR_Precharge_Press/(max(1,MAX
    _PRECHARGE_TIME-FR_Precharge_Timer));
  temp=min(2*TARGET_DUMP_RATE*LOOP_TIME,
    temp);
  temp=max(TARGET_DUMP_RATE*LOOP_TIME,
    temp);
  FL_Precharge_Press-=temp;
}
FR_Precharge_Press=max(0.0,FR_Precharge
  _Press);
}

```

```

else
{
FR_Precharge_Press=0.0;
}

```

**[00147]** In step 228 the old values for next loop are updated.

```

SS_Lin_Rear_Z1=RearLnrSlipAngle;
REAR_LNR_LAT_VEL=CurrentRearLnrLatVel;
FRONT_LNR_LAT_VEL=CurrentFrontLnrLatVel;
REAR_LNR_LAT_ACC=FltRearLnrLatA;
Z1_STEERING_WHEEL_ANGLE=STEERING_WHEEL_ANGLE;

```

#### PID CONTROLLER

**[00148]** Referring now to Figure 8, the PID controller 78 calculates front control pressures; using some or all of Proportional, Integral, Derivative and Double Derivative feedback control logic, required to control excessive two wheel lift during non-tripped rolling motion. For short the controller is referred to as a PID controller even though all the functions may not be provided. The "I" integral and "DD" double derivative functions are the most likely not to be present. When this controller acts on a signal it is referred to as PID control even though all the functions may not be provided. The control pressure is applied to the outer front wheel when the vehicle is experiencing high lateral acceleration while turning and concurrently experiencing wheel lift on the inner side of the turn. PID intervention is based on the requested PID feedback pressure, as well as vehicle velocity and lateral acceleration exceeding specified thresholds; when the vehicle is moving in a forward direction. Intervention ends when PID pressure falls below a threshold reflective of stabilization of roll motion, or if driver

is braking when PID pressure falls within an offset of driver pressure and slip of wheel being controlled indicates a level of stability appropriate for handoff to the anti-lock braking system.

**[00149]** Essentially, the PID controller 78 acts when more aggressive control is needed. The transition controller 76 acts before the vehicle is in an aggressive maneuver. This is typically below a threshold where the sensors are still in a linear region. Above the linear threshold, the PID takes over to aggressively control the vehicle. Aggressive control applies greater braking pressure to prevent the vehicle from rolling over.

**[00150]** In general, the Proportional, Integral, Derivative and Double Derivative, if present, terms are added together to form the total requested control pressure on each wheel. A brief explanation of each term follows.

**[00151]** The proportional term acts on a roll angle error input signal, thus the term proportional since the resulting requested control pressure will be proportional to the roll angle error magnitude by a factor  $K_P$ . A proportional peak hold strategy was added to mitigate a bouncing mode which can occur in certain aggressive maneuvers. The initiation of this strategy is contingent on the vehicle having experienced a recent divergence in roll rate magnitude.

**[00152]** The input to the derivative term is roll rate signal, which serves as a leading indicator of roll angle instability and thus provide an early lead on

controlling the transient behavior of roll angle. The gain factor  $K_D$  multiplies the roll rate signal minus a deadband to generate a derivative pressure term.  $KD$  is an experimentally derived term. If  $K_D$  is unduly high, unnecessary control interventions may be caused (sensitive system) and the system made susceptible to input signal noise.

**[00153]** The double derivative term is used to capture the roll stability tendency of the vehicle. The roll acceleration signal will exhibit wide-ranging oscillations during control, so the gain factor  $K_{DD}$  's influence on the overall PID stabilizing pressure should be set to a minimum.

**[00154]** The integral control pressure is used to drive the steady state roll angle error toward zero. A bounded version of the roll angle error signal is multiplied by a gain factor  $K_I$ , then integrated to provide a requested integrator pressure. The reason for bounding the input is to prevent integrator windup.

**[00155]** The PID controller 78 has various inputs. The inputs include a lateral acceleration at the center of gravity (CG\_FLT\_LAT\_ACC (m/s/s) input 300, a filtered roll rate (FLT\_ROLL\_RATE (deg/s) input 302, a (ROLL\_ACCELERATION (deg/s/s) input 304, (ROLL\_ANGLE\_TOTAL (deg) input 306, a (REFERENCE\_BANK\_ANGLE (deg) input 308, a front left brake pressure estimate (BRAKE\_PRESSR\_ESTMT [FL] (bar) input 312, a front right (BRAKE\_PRESSR\_ESTMT [FR] (bar) input 314, a driver requested pressure (DRIVER\_REQ\_PRESSURE (bar) 316, a (RSC\_REFERENCE\_VELOCITY (m/s) input 317, a (SLIP\_RATIO

[FL] (%) input 318, a (SLIP\_RATIO [FR] (%) input 320, (RIGHT\_TO\_LEFT\_TRANSITION (Boolean) input 322, (LEFT\_TO\_RIGHT\_TRANSITION (Boolean) input 324, (DRIVER BRAKING\_FLAG (Boolean) input 326, a roll system disabled (RSC\_DISABLED (Boolean) input 328, (REVERSE\_MOVEMENT (Boolean) input 330, (STATUS\_FIRST\_RUN (Boolean) input 332, (RSC\_IN\_CYCLE (Boolean) input 334, (AYC\_IN\_CYCLE (Boolean) input 336 and a roll signal for control input 338.

**[00156]** The PID controller has various outputs including (PID\_ACTIVE [FL] (Boolean) output 350, (PID\_ACTIVE [FR] (Boolean) output 352, a stabilizer pressure (PID\_STBLZ\_PRESSURE [FL] (bar) output 354, (PID\_STBLZ\_PRESSURE [FR] (bar) output 356, (RECENT\_AYC\_CNTRL\_EVENT (Boolean) output 358, (RECENT\_PID\_CNTRL\_EVENT (Boolean) output 360, (INCREASE\_REQUESTED\_PRESSURE [FL] (Boolean) output 362, (INCREASE\_REQUESTED\_PRESSURE [FR] (Boolean) output 364.

**[00157]** The PID controller 78 includes the various calibratable parameters. The parameters are shown for his example. The parameters in implementation may be varied based on the vehicle configuration. Proportional dead band PROP\_DB (linear interpolation function of vehicle speed): Absolute value of Proportional deadband above which the error signal input to the proportional controller becomes positive, thus yielding a positive requested proportional pressure. At nominal speeds, the value is chosen in the vicinity of the vehicle roll gradient experienced near 1 g of lateral acceleration.

RSC_REFERENCE_VELOCITY (m/s)	PROP_DB (deg)
0	7.0
16.0	5.2
32.0	5.2
83.0	4.6

**[00158]** Another is proportional gain factor  $K_{P\_UP}$  (linear interpolation function of vehicle speed): Proportional gain factor that multiplies the roll angle error signal when above PROP\_DB, thus generating a positive proportional term of the requested PID stabilizing pressure.

RSC_REFERENCE_VELOCITY (m/s)	$K_{P\_UP}$ (bar/deg)
0	30.0
18.0	30.0
35.0	30.0
83.0	30.0

**[00159]** Proportional gain factor  $K_{P\_DOWN}$  (linear interpolation function of vehicle speed): Proportional gain factor that multiplies the roll angle error signal when below PROP\_DB, thus generating a negative proportional term of the requested PID stabilizing pressure.

RSC_REFERENCE_VELOCITY (m/s)	$K_{P\_DOWN}$ (bar/deg)
---------------------------------	----------------------------

0	40.0
18.0	40.0
35.0	40.0
83.0	40.0

**[00160]**    `LARGE_ROLL_RATE_THRESH` (in this example a value of 30.0 deg/s is used): Roll rate magnitude above which recent large roll rate timer is set to the maximum value.

**[00161]**    `RECENT_LARGE_ROLL_RATE` (0.5 sec): Time duration assigned to recent large roll rate timer when roll rate magnitude exceeds `LARGE_ROLL_RATE_THRESH`.

**[00162]**    `PROP_HOLD_ANGLE_THRESHOLD` (in this example a value of  $1.48 \times \text{Roll\_gradient}$  is used): Roll angle at which the proportional pressure starts tracking peak proportional pressure.

**[00163]**    `PROP_PEAK_HOLD_TIME` (in this example a value of 0.5 sec is used): Time duration that the proportional peak hold pressure is latched once magnitude of roll signal for control falls below `PROP_HOLD_ANGLE_THRESHOLD`.

**[00164]**    `PROP_PEAK_RAMP_DOWN_RATE` (in this example a value of 250 bar/s is used): Ramp down rate of prior proportional peak to a new lower prop peak pressure when roll signal for control still exceeds hold angle, or ramp down rate to upstream proportional pressure if roll signal for control has not exceeded hold angle within the last `PROP_PEAK_HOLD_TIME` seconds.

**[00165]**    DERIV\_DB    (linear interpolation function of lateral acceleration):    Absolute value of derivative deadband, used outside of PID control, above which the error signal input to the derivative controller becomes positive, thus yielding a positive requested derivative pressure.    Values are chosen to prevent roll rate noise at low lateral acceleration levels from inducing nuisance PID interventions.

CG_FLT_LAT_ACC (m/s/s)	DERIV_DB (deg/sec)
0	5.0
4.0	5.0
7.0	0.0
100.0	0.0

**[00166]**    DERIV\_DB\_DURING\_PID\_CONTROL (in this example a value of -20 deg/s is used).    During PID intervention the deadband is set to a constant (non-lateral acceleration dependent) negative value, providing a phase advance on the PID pressure to mitigate roll oscillation during aggressive maneuvers.

**[00167]**    K\_D\_UP    (linear interpolation function of vehicle speed):    Derivative gain factor that multiplies the roll rate error signal when roll rate is above DERIV\_DB; generating a positive derivative term of the requested PID stabilizing pressure.



RSC_REFERENCE_VELOCITY (m/s)	K_D_UP (bar/deg/s)
0	1.0
18.0	1.0
35.0	1.0
83.0	1.0

**[00168]** K\_D\_DOWN (linear interpolation function of vehicle speed): Derivative gain factor that multiplies the roll rate error signal when roll rate is below DERIV\_DB; generating a negative derivative term of the requested PID stabilizing pressure.

RSC_REFERENCE_VELOCITY (m/s)	K_D_DOWN (bar/deg/s)
0	0.2
18.0	0.2
35.0	0.2
83.0	0.2

**[00169]** K\_DD\_UP (in this example a value of 0. bar/deg/s/s is used): Double derivative gain factor that multiplies the roll acceleration signal when RSC intervention is active and roll acceleration sign matches that of the turn; generating a positive double derivative term of the requested PID stabilizing pressure.

**[00170]** K\_DD\_DOWN (in this example a value of 0 bar/deg/s/s is used): Double derivative gain factor

that multiplies the roll acceleration signal when RSC intervention is not active or roll acceleration sign is opposite that of the turn; generating a nil double derivative term of the requested PID stabilizing pressure.

**[00171]** INTG\_DB (linear interpolation function of vehicle speed): Absolute value of Integral deadband above which the error signal input to the Integral term becomes positive. At nominal speeds, value is chosen in the vicinity of the vehicle roll gradient experienced near 1 g of lateral acceleration.

RSC_REFERENCE_VELOCITY (m/s)	INTG_DB (deg)
0	7.0
16.0	5.2
32.0	5.2
83.0	4.6

**[00172]** MAX\_INTGRTR\_ERROR (in this example a value of 5.0 deg is used): Absolute value of upper bound on the error signal input to the Integral controller. The purpose of this parameter is to prevent integrator windup.

**[00173]** MIN\_INTGRTR\_ERROR (in this example a value of 5.0 deg is used): Absolute value of lower bound on the error signal input to the Integral controller. The purpose of this parameter is to avoid integrator windup.

**[00174]** K\_I (in this example a value of 10.0 bar/deg.s is used): Integral gain factor multiplying the bounded

roll angle error signal times loop time, generating the integral term of the requested stabilizing pressure.

**[00175]**    `PRESSURE_INCREASE_DELTA_FOR_INFLECTION_ADJUST`  
(in this example a value of 10 bar is used): The pressure delta that the pressure estimate has to increase by before the downward adjustment of the PID pressure, to an offset (`PRESSURE_OFFSET_DURING_PID_RAMP_UP`) from the pressure estimate, occurs.

**[00176]**    `PRESSURE_OFFSET_DURING_PID_RAMP_UP` (in this example a value of 20 bar is used): Specifies the maximum delta between PID pressure and estimated pressure, once the inflection adjustment begins.

**[00177]**    `LMTD_RAMP_DOWN_RATE` (in this example a value of -400 bar/s is used): Maximum decrease rate that the PID requested pressure is allowed to ramp down at.

**[00178]**    `MAX_PRESSURE_DECREASE_TO_ENTER_LMTD_RAMP_DOWN`  
(in this example a value of -0.5 bar is used): Negative delta pressure of the underlying P+I+D+DD sum required to enter the limited PID ramp down mode.

**[00179]**    `MIN_PRESSURE_INCREASE_TO_STAY_IN_LMTD_RAMP_DOWN`  
N (in this example a value of 0.5 bar is used): Positive delta pressure of the underlying P+I+D+DD sum above which limited PID ramp down mode exits.

**[00180]**    `MAX_RAMP_DOWN_RATE` (in this example a value of -300 bar/s is used): Ramp down rate used when opposite front wheel requests PID intervention concurrently as inside wheel is ramping down PID requested pressure.

**[00181]** ENTER\_THRES (in this example a value of 30.0 bar is used): PID requested pressure threshold for either front wheel, above which the flag PID\_ACTIVE is set. A true value for the aforementioned flag constitutes one of the necessary conditions for activating RSC.

**[00182]** LAT\_ACC\_ACTVTION\_THRSHLD (in this example a value of 5.0 m/s/s is used): Absolute value of vehicle lateral acceleration threshold above which the variable LARGE\_LAT\_ACC\_COUNTER is initialized to LAT\_ACC\_COUNTER\_INIT. A nonzero value of this counter is a necessary condition for activating RSC.

**[00183]** LAT\_ACC\_COUNTER\_INIT (in this example a value of 3.0 sec is used): Time duration assigned to LARGE\_LAT\_ACC\_COUNTER when vehicle lateral acceleration magnitude exceeds LARGE\_ROLL\_RATE\_THRESH.

**[00184]** EXIT\_THRES (in this example a value of 10.0 bar is used): PID requested pressure threshold for either front wheel below which the flag PID\_ACTIVE resets to false, thus exiting PID control.

**[00185]** WHEEL\_STABLE\_IN\_SLIP (in this example a value of -15 % is ;used): Maximum slip ratio observed on PID controlled wheel before PID intervention can exit.

**[00186]** MIN\_PID\_PRES\_FOR\_FORCED\_CONTROL\_EXIT (in this example a value of 9 bar is used): PID pressure threshold below which PID intervention is forced to exit, in case wheel is allowed to lock for an extended period thus not allowing the WHEEL\_STABLE\_IN\_SLIP criteria to be met.

**[00187]** RECENT\_PID\_EVNT\_THRSHLD (in this example a value of 0.7 sec is used): Period of time during which the history of any PID intervention is logged.

**[00188]** RECENT\_AYC\_EVNT\_THRSHLD (in this example a value of 0.7 sec is used): Period of time during which the history of any AYC intervention is logged.

**[00189]** PID\_MINIMUM\_ACTIVATION\_SPEED (in this example a value of 7.0 m/s is used): Vehicle speed below which RSC system will not activate.

**[00190]** MAXIMUM\_SPEED\_TO\_CONTINUE\_PID (in this example a value of 5.0 m/s is used: If vehicle speed falls below this threshold during PID intervention, PID control exits.

**[00191]** MAXIMUM\_STBLZ\_PRESSURE (in this example a value of 255.0 bar is used).

**[00192]** LOOP\_TIME\_SEC: (in this example a value of 0.007 sec is used): Sampling time of input signals, as well as maximum allowable execution time of stability control system logic.

#### Compute PID Desired Braking Pressures Logic

**[00193]** The description of the PID feedback controller for non-tripped roll events follows. An explanation of the calculation of each term is included, and a C-language implementation of the computation is also included.

**[00194]** Referring now to Figure 9, a flowchart illustrating the operation of the PID controller 78 is illustrated. In step 400 the sensors of the system are

read. In step 402 the various inputs described above are obtained.

**[00195]** In step 404 a derivative term of the roll stabilizing requested pressures for each front wheel is determined. During PID intervention or if a recent moderately aggressive transition maneuver (as indicated from the transition controller), the derivative deadband determined in step 406 is set to a constant negative threshold (DERIV\_DB\_DURING\_PID\_CONTROL) to provide for a prediction of normal load oscillations. Otherwise the derivative deadband (DERIV\_DB) is a function of the vehicle's lateral acceleration, nearing zero as the lateral acceleration increases beyond 0.7 g's.

**[00196]** For left turns (positive roll angle), compute derivative pressure term for outer (right) front wheel.

```
if (PID_ACTIVE[FR]||RIGHT_TO_LEFT_TRANSITION)
    derivative_db=DERIV_DB_DURING_PID_CONTROL;
else
    derivative_db=DERIV_DB;

if ((FLT_ROLL_RATE-derivative_db)>0) DrvtvPres
    [FR]=K_D_UP*(FLT_ROLL_RATE-derivative_db);
else
    DrvtvPres[FR]=K_D_DOWN*(FLT_ROLL_RATE-
        Derivative_db);
```

**[00197]** For right turns (negative roll angle), compute derivative pressure term for outer (left) front wheel.

```
if (PID_ACTIVE[FL]||LEFT_TO_RIGHT_TRANSITION)
    derivative_db=DERIV_DB_DURING_PID_CONTROL;
else
    derivative_db=DERIV_DB;

if ((FLT_ROLL_RATE+derivative_db)<0) DrvtvPres[FL]=
    (-1.0)*K_D_UP*(FLT_ROLL_RATE+derivative_db);
else
```

DrvtvPres[FL]=(-  
1.0)\*K\_D\_DOWN\*(FLT\_ROLL\_RATE+derivative\_db);

**[00198]** In step 406, the double derivative term of the roll stabilizing requested pressures for each front wheel, based on roll acceleration signal is determined. This term is effective during RSC intervention, as K\_DD\_DOWN is set to zero.

**[00199]** For left turns (positive roll angle), the double derivative pressure term for outer (right) front wheel is determined.

```
if ((ROLL_ACCELERATION>0)&&RSC_IN_CYCLE)DDPres[FR]  
    =K_DD_UP*ROLL_ACCELERATION;  
else  
    DDPres[FR]=K_DD_DOWN*ROLL_ACCELERATION;
```

**[00200]** For right turns (negative roll angle), compute derivative pressure term for outer (left) front wheel.

```
if ((ROLL_ACCELERATION<0)&&RSC_IN_CYCLE)DDPres[FL]  
    =(-1.0)*K_DD_UP*ROLL_ACCELERATION;  
else  
    DDPres[FL]=(-1.0)*K_DD_DOWN*ROLL_ACCELERATION;
```

**[00201]** In step 408 a proportional term of the roll stabilizing requested pressures for each front wheel is determined. A roll signal for control is determined based on the vehicle angle. In the present example the ROLL\_SIG\_FOR\_CONTROL is based on the difference between ROLL\_ANGLE\_TOTAL and REFERENCE\_BANK\_ANGLE as set forth in:

```
ROLL_SIG_FOR_CONTROL=ROLL_ANGLE_TOTAL-REFERENCE  
    _BANK_ANGLE;
```

**[00202]** The proportional term is based on roll angle error signal formed by subtracting a deadband, of value

PROP\_DB, from the input control signal ROLL\_SIG\_FOR\_CONTROL. This error signal is then multiplied by the proportional gain factor K\_P to obtain the proportional pressure term.

**[00203]** For left turns (positive roll angle), compute proportional pressure term for outer (right) front wheel.

```
if ((ROLL_SIG_FOR_CONTROL-PROP_DB)>0) PropPres[FR]
    =(ROLL_SIG_FOR_CONTROL-PROP_DB)*K_P_UP;
else
    PropPres[FR]=(ROLL_SIG_FOR_CONTROL-PROP_DB)
        *K_P_DOWN;
```

**[00204]** A history of recent large positive roll rate is kept. The history is used as a screening criteria for initiating proportional peak hold logic described thereafter. RECENT\_LARGE\_PSTV\_ROLL\_RATE\_TIMER is used to keep track of this criteria.

```
if (FLT_ROLL_RATE>LARGE_ROLL_RATE_THRESH)
{
    RECENT_LARGE_PSTV_ROLL_RATE_TIMER=
        RECENT_LARGE_ROLL_RATE;
}
else if (RECENT_LARGE_PSTV_ROLL_RATE_TIMER>0)
{
    RECENT_LARGE_PSTV_ROLL_RATE_TIMER--;
}
if (RECENT_LARGE_PSTV_ROLL_RATE_TIMER>0) RECENT
    _LARGE_PSTV_ROLL_RATE=TRUE;
else
    RECENT_LARGE_PSTV_ROLL_RATE=FALSE;
```

**[00205]** Proportional peak hold logic: When a large positive roll divergence starts building, as indicated by roll signal for control exceeding PROP\_HOLD\_ANGLE\_THRESHOLD and corroborated by the existence of a recent large positive roll rate, a



constant base level of PID pressure to achieve a consistent level of deep slip on the wheel is maintained. This helps mitigate vehicle bounce during aggressive maneuvers. The constant base level is obtained by holding the peak value of the proportional term until the next proportional peak that exceeds the threshold, or until a timer (PROP\_PEAK\_HOLD\_TIMER[FR]) runs out.

**[00206]** Figure 10 depicts the intended proportional peak hold strategy. In region A the proportional signal is held. In region B, because the new proportional peak pressure is increasing, the proportional hold pressure is allowed to track to a new peak. In region C the proportional pressure decreases, so the hold pressure is ramped down. In Region D the hold pressure continues to ramp down since the proportional pressure continues to decrease. It should be noted that the proportional peak hold value is held at least until the timer is reset. The countdown timer is maintained at its peak value until the roll signal for control drops below the proportional hold angle threshold.

**[00207]** This is set forth in the following logic:

```
if ((ROLL_SIG_FOR_CONTROL>PROP_HOLD_ANGLE
    _THRESHOLD)
    &&(RECENT_LARGE_PSTV_ROLL_RATE||(PROP_PEAK
        _HOLD_TIMER[FR]>0)))
{ //roll angle exceeds threshold AND recent large
  positive roll rate was seen or have already entered
  prop peak hold mode, //enter/keep in prop peak hold
  mode

if (PROP_PEAK_HOLD_TIMER[FR]==0)
{ //first divergent oscillation indicated
  by a null prop hold timer
```

```

INITIAL_DIVERGENT_OSCILLATION[FR]=TRUE;
}
if (INITIAL_DIVERGENT_OSCILLATION[FR])
{
    //prop hold logic for first divergent
    oscillation
    PROP_PEAK_HOLD_TIMER[FR]=PROP_HOLD_LOOPS;
    //set timer to max value during divergence
    if (PropPres[FR]>=REFERENCE_PROP_PEAK_HOLD
        [FR])
    {
        //PropPres is still rising, let PeakHold
        track it
        REFERENCE_PROP_PEAK_HOLD[FR]=PropPres
        [FR];
        PROP_PEAK_HOLD[FR]=FALSE;
    }
    else
    {
        //PropPres is no longer rising, assign latched
        PeakHold to PropPres
        PropPres[FR]=REFERENCE_PROP_PEAK_HOLD[FR];
        PROP_PEAK_HOLD[FR]=TRUE;
    }
}
else //prop hold logic for subsequent divergent
oscillations
{
    PROP_PEAK_HOLD_TIMER[FR]=PROP_HOLD_LOOPS;//set
    timer to max value during
    divergence//Logic to handle transitions
    to a higher or lower subsequent
    proportional peak roll signal

    if (PropPres[FR]>NEW_PROP_PEAK_HOLD[FR])
    {
        //Prop Pres is still rising, assign a
        NewPeakHold to track it
        NEW_PROP_PEAK_HOLD[FR]=PropPres[FR];
        TRACKING_NEXT_PEAK[FR]=TRUE;//flag
        indicating NewPeakHold is in tracking
        mode RAMP_DOWN_TO_NEW_PROP_PEAK[FR]
        =FALSE; //reset ramp down once tracking
        peak starts
    }
    else
    {
        //PropPres is no longer rising, reset
        tracking flag
        TRACKING_NEXT_PEAK[FR]=FALSE;
    }
    //Logic to handle transition from Reference
    PeakHold to NewPeakHold

```

```

        if
        ((PropPres[FR]>REFERENCE_PROP_PEAK_HOLD[FR])&&
         TRACKING_NEXT_PEAK[FR])
        {
            //allow PropPres increase above prior PeakHold
            during tracking mode
            PROP_PEAK_HOLD[FR]=FALSE;
        }
    else if ((PropPres[FR]>REFERENCE_PROP_PEAK_HOLD
             [FR])&&!TRACKING_NEXT_PEAK[FR])
    {
        //NewPeakHold is latched and is > prior
        PeakHold, so REFERENCE_PROP_PEAK_HOLD[FR]=NEW
        _PROP_PEAK_HOLD[FR]; //assign NewPeakHold

//to PeakHold, and
        PropPres[FR]=REFERENCE_PROP_PEAK_HOLD[FR]
        //assign latched NewPeakHold to PropPres
        RAMP_DOWN_TO_NEW_PROP_PEAK[FR]=FALSE;
        //since ramp up to NewPeakHold occurred,
        //ramp down is not active
        PROP_PEAK_HOLD[FR]=TRUE;
    }
    else if ((PropPres[FR]<REFERENCE_PROP_PEAK
             _HOLD[FR]))
    {
        if (!TRACKING_NEXT_PEAK[FR]
            &&(REFERENCE_PROP_PEAK_HOLD[FR]>NEW_PROP
              _PEAK_HOLD[FR]))
        {
            //NewPeakHold is latched and is<prior
            PeakHold, REFERENCE_PROP_PEAK_HOLD[FR]
            -=PROP_PEAK_RAMP_DOWN_RATE*LOOP_TIME_SEC;
            //ramp down prior PeakHold toNewPeak
            Hold,RAMP_DOWN_TO_NEW_PROP_PEAK[FR]=TRUE;

//indicate ramp down is in effect

            if (REFERENCE_PROP_PEAK_HOLD[FR]<NEW
              _PROP_PEAK_HOLD[FR])
        {
            //ramp down of prior PeakHold brings it
            to NewPeakHold level, // assign NewPeak
            Hold to PeakHold, and REFERENCE_PROP_PEAK
            _HOLD[FR]=NEW_PROP_PEAK_HOLD[FR];

            RAMP_DOWN_TO_NEW_PROP_PEAK[FR]=FALSE;
            //indicate end of ramp down//of PeakHold
            towards NewPeakHold
        }
    }
    PropPres[FR]=REFERENCE_PROP_PEAK_HOLD[FR]; //assign

```

```

PeakHold to PropPresPROP_PEAK_HOLD[FR]=TRUE;
    }
}
else
{
    //roll angle is <=threshold, so decrement
    timer and assign PropPres to PeakHold if
    (PROP_PEAK_HOLD_TIMER[FR]>1)
    {
        //PropHold timer is>1, therefore hold
        PropPres INITIAL_DIVERGENT
        _OSCILLATION[FR]=FALSE; //first divergent
        oscillation of prop peak hold mode has
        ended PROP_PEAK_HOLD_TIMER[FR]--;
        //decrement PropHold timer
        if (RAMP_DOWN_TO_NEW_PROP_PEAK[FR])
    {
        //ramp down of prior PeakHold to NewPeakHold
        is not completed, continue till done
        REFERENCE_PROP_PEAK_HOLD[FR]-PROP_PEAK
        _RAMP_DOWN_RATE*LOOP_TIME_SEC;
    {
        //ramp down prior PeakHold to NewPeakHold
        if (REFERENCE_PROP_PEAK_HOLD[FR]<NEW
        _PROP_PEAK_HOLD[FR])
        {
            //ramp down of prior PeakHold brings
            it to NewPeakHold level,
            //assign NewPeakHold to PeakHold
            REFERENCE_PROP_PEAK_HOLD[FR]=NEW
            _PROP_PEAK_HOLD[FR];

            RAMP_DOWN_TO_NEW_PROP_PEAK[FR]
            =FALSE;//indicate end of ramp down
            of
            //PeakHold towards NewPeakHold
            }
        }
        else
        {
            //ramp down from old to new PeakHold
            has ended or was not in effect, so reset
            NewPeakHold state
            NEW_PROP_PEAK_HOLD[FR]=0;
        }
    }
    if (PropPres[FR]<REFERENCE_PROP_PEAK_HOLD[FR])
    { //overwrite decreasing PropPres with
    PeakHold during peak hold mode
        PropPres[FR]=REFERENCE_PROP_PEAK_HOLD
        [FR];
        PROP_PEAK_HOLD[FR]=TRUE;
    }
    else

```

```

        {
            //PropPres is higher than PeakHold, allow
            PropPres increase above PeakHold by
            PROP_PEAK_HOLD[FR]=FALSE; //exiting prop
            hold mode and keeping upstream PropPres
        }
    }
else //PropHold timer<=1, ramp out of prop PeakHold
    if it existed //otherwise upstream PropPres
    value is kept unchanged
    {
        if (PROP_PEAK_HOLD_TIMER[FR]==1)
        {
            //prop hold mode was in effect, set flag
            indicating ramp down of PeakHold to PropPres
            RAMP_DOWN_PROP_HOLD_TO_PROP_PRES[FR]
            =TRUE;
            PROP_PEAK_HOLD_TIMER[FR]=0;
        }
        if (RAMP_DOWN_PROP_HOLD_TO_PROP_PRES[FR])
        {
            //ramp down of PeakHold to PropPres
            if (REFERENCE_PROP_PEAK_HOLD[FR]>(PropPres[FR]
                +PROP_PEAK_RAMP_DOWN_RATE*LOOP_TIME_SEC))
            {
                REFERENCE_PROP_PEAK_HOLD[FR]-=PROP_PEAK
                _RAMP_DOWN_RATE*LOOP_TIME_SEC;//ramp
                down prior PeakHold to NewPeakHold
                PropPres[FR]=REFERENCE_PROP_PEAK
                _HOLD[FR];
            }
            else //PeakHold reached upstream PropPres, end
            ramp down and exit prop hold logic
            {
                REFERENCE_PROP_PEAK_HOLD[FR]=0;
                RAMP_DOWN_PROP_HOLD_TO_PROP_PRES[FR]
                =FALSE;
            }
        }
    }
else //ramp down of PeakHold to PropPres has ended,
or
{
    //roll angle never exceeded threshold,
    therefore REFERENCE_PROP_PEAK_HOLD[FR]=0;//reset
    reference prop peak pressure
}
NEW_PROP_PEAK_HOLD[FR]=0;
PROP_PEAK_HOLD[FR]=FALSE;
RAMP_DOWN_TO_NEW_PROP_PEAK[FR]=FALSE;
}
}

```

**[00208]** For right turns (negative roll angle), compute proportional pressure term for outer (left) front wheel by:

```
if ((ROLL_SIG_FOR_CONTROL+PROP_DB)<0)
    PropPres[FL]=(-1.0)*(ROLL_SIG_FOR_CONTROL+PROP
        _DB)*K_P_UP;
else
    PropPres[FL]=(-1.0)*(ROLL_SIG_FOR_CONTROL+PROP
        _DB)*K_P_DOWN;
```

**[00209]** Keep history of recent large negative roll rate, which is used as a screening criteria for initiating proportional peak hold logic thereafter. RECENT\_LARGE\_NGTV\_ROLL\_RATE\_TIMER is used to keep track of this criteria.

```
if (FLT_ROLL_RATE<(-1*LARGE_ROLL_RATE_THRESH))
{
    RECENT_LARGE_NGTV_ROLL_RATE_TIMER=RECENT
        _LARGE_ROLL_RATE;
}
else if (RECENT_LARGE_NGTV_ROLL_RATE_TIMER>0)
{
    RECENT_LARGE_NGTV_ROLL_RATE_TIMER--;
}
if (RECENT_LARGE_NGTV_ROLL_RATE_TIMER>0)RECENT
    _LARGE_NGTV_ROLL_RATE=TRUE;
else
    RECENT_LARGE_NGTV_ROLL_RATE=FALSE;
```

**[00210]** When a large negative roll divergence starts building, as indicated by roll signal for control exceeding PROP\_HOLD\_ANGLE\_THRESHOLD and corroborated by the existence of a recent large negative roll rate, this strategy serves to keep a constant base level of PID pressure to achieve a consistent level of deep slip on the wheel and thus help mitigate vehicle bounce during aggressive maneuvers. The constant base level is obtained by holding the peak value of the proportional

term until the next proportional peak that exceeds the threshold, or until PROP\_PEAK\_HOLD\_TIMER[FL] runs out.

```
if ((ROLL_SIG_FOR_CONTROL<(-1.0*PROP_HOLD_ANGLE
    _THRESHOLD))
    &&(RECENT_LARGE_NGTV_ROLL_RATE||(PROP_PEAK
    _HOLD_TIMER[FL]>0))
)
{
    //roll angle exceeds threshold AND recent
    large negative roll rate was seen or have already
    entered prop peak hold mode,
    //enter/keep in prop peak hold mode

    if (PROP_PEAK_HOLD_TIMER[FL]==0)
    {
        //first divergent oscillation indicated
        by a null prop hold timer
        INITIAL_DIVERGENT_OSCILLATION[FL]=TRUE;
    }
    if (INITIAL_DIVERGENT_OSCILLATION[FL])
    {
        //prop hold logic for first divergent
        oscillation
        PROP_PEAK_HOLD_TIMER[FL]=PROP_HOLD_LOOPS;
        //set timer to max value during
        divergence
        if (PropPres[FL]>=REFERENCE_PROP_PEAK_HOLD
            [FL])
        {
            //PropPres is still rising, let PeakHold
            track it
            REFERENCE_PROP_PEAK_HOLD[FL]=PropPres
            [FL];
            PROP_PEAK_HOLD[FL]=FALSE;
        }
        else
        {
            //PropPres is no longer rising, assign
            latched PeakHold to PropPres
            PropPres[FL]=REFERENCE_PROP_PEAK_HOLD
            [FL];
            PROP_PEAK_HOLD[FL]=TRUE;
        }
    }
    else
        //prop hold logic for subsequent
        divergent oscillations

    {
        PROP_PEAK_HOLD_TIMER[FL]=PROP_HOLD_LOOPS;
        //set timer to max value during divergence
    }
}
```

```

        //Logic to handle transitions to a higher
or lower subsequent proportional peak roll
signal

        if (PropPres[FL]>NEW_PROP_PEAK_HOLD[FL])
    {
        //PropPres is still rising, assign a
        NewPeakHold to track it
        NEW_PROP_PEAK_HOLD[FL]=PropPres[FL];
        TRACKING_NEXT_PEAK[FL]=TRUE;//flag
        indicating NewPeakHold is in tracking
        mode
        RAMP_DOWN_TO_NEW_PROP_PEAK[FL]=FALSE;
        //reset ramp down once tracking peak
        starts
    }
else
    {
        //PropPres is no longer rising, reset tracking
        flag
        TRACKING_NEXT_PEAK[FL]=FALSE;
    }
//Logic to handle transition from ReferencePeakHold
to NewPeakHold

    if
    ((PropPres[FL]>REFERENCE_PROP_PEAK_HOLD[FL])
    &&TRACKING_NEXT_PEAK[FL])
    {
        //allow PropPres increase above prior PeakHold
        during tracking modePROP_PEAK_HOLD[FL]=FALSE;
    }
else if ((PropPres[FL]>REFERENCE_PROP_PEAK
        _HOLD[FL])&&!TRACKING_NEXT_PEAK[FL])
    {
        //NewPeakHold is latched and is>prior
        PeakHold, so REFERENCE_PROP_PEAK_HOLD[FL]=NEW
        _PROP_PEAK_HOLD[FL];
        //assign NewPeakHold
        //to PeakHold, and PropPres[FL]
        =REFERENCE_PROP_PEAK_HOLD[FL];

        //assign latched NewPeakHold to PropPres
        RAMP_DOWN_TO_NEW_PROP_PEAK[FL] =FALSE;
        //since ramp up to NewPeakHold occurred,
        // ramp down is not active PROP_PEAK_HOLD
        [FL]=TRUE;
    }
else if ((PropPres[FL]<REFERENCE_PROP_PEAK_HOLD
        [FL]))
    {

```



```

if (!TRACKING_NEXT_PEAK[FL]&&(REFERENCE_PROP
    _PEAK_HOLD[FL]>NEW_PROP_PEAK_HOLD[FL]))

{
    //NewPeakHold is latched and is<prior
    PeakHold, REFERENCE_PROP_PEAK_HOLD[FL]-=PROP_PEAK
    _RAMP_DOWN_RATE*LOOP_TIME_SEC;
    //ramp down prior PeakHold to
    NewPeakHold, RAMP_DOWN_TO_NEW_PROP_PEAK[FL]=TRUE;
    //indicate ramp down is in effect
    if (REFERENCE_PROP_PEAK_HOLD[FL]<NEW_PROP_PEAK
        HOLD[FL])
    {
        //ramp down of prior PeakHold brings it to
        NewPeakHold level, //assign NewPeakHold to
        PeakHold, and REFERENCE_PROP_PEAK_HOLD[FL]=NEW
        _PROP_PEAK_HOLD[FL];

        RAMP_DOWN_TO_NEW_PROP_PEAK[FL]=FALSE; //indicate
        end of ramp down
        // of PeakHold towards NewPeakHold
    }
}
PropPres[FL]=REFERENCE_PROP_PEAK_HOLD[FL];
    //assign PeakHold to PropPres
    PROP_PEAK_HOLD[FL]=TRUE;
}
}
}
else
{
    //roll angle is<=threshold, so decrement timer
    and assign PropPres to PeakHold
    if (PROP_PEAK_HOLD_TIMER[FL]>1)

    {
        //PropHold timer is > 1, therefore hold
        PropPres

        INITIAL_DIVERGENT_OSCILLATION[FL]=FALSE;
        //first divergent oscillation of prop peak hold
        mode has ended PROP_PEAK_HOLD_TIMER[FL]--;
        //decrement PropHold timer

        if (RAMP_DOWN_TO_NEW_PROP_PEAK[FL])
        {
            //ramp down of prior PeakHold to NewPeakHold
            is not completed, continue till done
            REFERENCE_PROP_PEAK_HOLD[FL]-=PROP
            _PEAK_RAMP_DOWN_RATE*LOOP_TIME_SEC;
            // ramp down prior PeakHold to NewPeakHold

            if (REFERENCE_PROP_PEAK_HOLD[FL]<NEW_PROP

```

```

    _PEAK_HOLD[FL])
{
    //ramp down of prior PeakHold brings it to
    NewPeakHold level,
    //assign NewPeakHold to PeakHoldREFERENCE
    _PROP_PEAK_HOLD[FL]=NEW_PROP_PEAK_HOLD[FL];

    RAMP_DOWN_TO_NEW_PROP_PEAK[FL]=FALSE;    //indicate
    end of ramp down of //PeakHold towards NewPeakHold
}
}
else
{
    //ramp down from old to new PeakHold has
    ended or was not in effect, so reset New
    PeakHold stateNEW_PROP_PEAK_HOLD
    [FL]=0;
}
if (PropPres[FL]<REFERENCE_PROP_PEAK_HOLD[FL])
{
    //overwrite decreasing PropPres with PeakHold
    during peak hold mode PropPres[FL]=REFERENCE_PROP
    _PEAK_HOLD[FL];PROP_PEAK_HOLD[FL]=TRUE;
}
else
{
    //PropPres is higher than PeakHold, allow
    PropPres increase above PeakHold by
    PROP_PEAK_HOLD[FL]= FALSE;
    //exiting prop hold mode and keeping upstream
    PropPres
}
}
else //PropHold timer<=1, ramp out of prop PeakHold
if it existed

//otherwise upstream PropPres value is kept
unchanged

{

if (PROP_PEAK_HOLD_TIMER[FL]==1)

{
    //prop hold mode was in effect, set flag
    indicating ramp down of PeakHold to PropPres

    RAMP_DOWN_PROP_HOLD_TO_PROP_PRES[FL]=TRUE;
    PROP_PEAK_HOLD_TIMER[FL]=0;
}
if (RAMP_DOWN_PROP_HOLD_TO_PROP_PRES[FL])
{
    //ramp down of PeakHold to PropPres

```

```

if (REFERENCE_PROP_PEAK_HOLD[FL]>(PropPres[FL]
+PROP_PEAK_RAMP_DOWN_RATE*LOOP_TIME_SEC))
{
    REFERENCE_PROP_PEAK_HOLD[FL]-=PROP
        _PEAK_RAMP_DOWN_RATE*LOOP_TIME_SEC;
//ramp down prior PeakHold to NewPeakHoldPropPres
    [FL]=REFERENCE_PROP_PEAK_HOLD[FL];
}
else //PeakHold reached upstream PropPres, end
ramp down and exit prop hold logic
{
    REFERENCE_PROP_PEAK_HOLD[FL]=0;
    RAMP_DOWN_PROP_HOLD_TO_PROP_PRES[FL]=FALSE;
}
}
else //ramp down of PeakHold to PropPres has ended,
or
{
    //roll angle never exceeded threshold,
    therefore REFERENCE_PROP_PEAK_HOLD[FL]=0;
    //reset reference prop peak pressure
}
    NEW_PROP_PEAK_HOLD[FL]=0;
    PROP_PEAK_HOLD[FL]=FALSE;
    RAMP_DOWN_TO_NEW_PROP_PEAK[FL]=FALSE;
}
}

```

**[00211]** Referring back to Figure 9, in step 412 an intermediate sum of the P, D and DD pressures for front left and front right wheels may be obtained as follows:

```

//Calculate current loop's value of P_D_DD sum for
front right wheel P_D_DD_Sum [FR]=PropPres[FR]
+DrvTvPres[FR]+DDPres[FR];

//Calculate current loop's value of P_D_DD sum for
front left wheel P_D_DD_Sum [FL]=PropPres[FL]
+DrvTvPres[FL]+DDPres[FL];

```

**[00212]** In step 414 an integral term of the roll stabilizing requested pressures for each front wheel is determined. A deadband, of value INTG\_DB, is subtracted from ROLL\_SIG\_FOR\_CONTROL to form the input error signal to the integral pressure term.

**[00213]** For the integral term, special precaution has to be taken regarding integral windup. Integrator windup is a condition that occurs when the input error signal remains large; which in turn causes the integrator term to grow (wind up) to very large values. Even after the roll angle falls within the target deadband, the integrator term takes an excessively long time to unwind; thus continuing to needlessly command control pressure. In order to avoid this situation, the following steps are applied: Impose an upper and lower bound on the input error signal [ROLL\_SIG\_FOR\_CONTROL - INTG \_DB], defined by MAX\_INTGRTR\_ERROR and MIN\_INTGRTR\_ERROR respectively. This limits the error term that gets added (integrated) each execution loop and compute an integral pressure term based on this bound input error signal.

**[00214]** For left turns (positive roll angle), the integral pressure term for outer (right) front wheel is determined by:

```
//Compute raw value of Integral error for right
wheel IntegralError[FR]=ROLL_SIG_FOR_CONTROL
-INTG_DB;

//Put bounds on min and max values of the integral
error, to prevent integrator windup IntegralError
[FR]=LIMIT (IntegralError[FR], ((-1)*MIN_INTGRTR
_ERROR), MAX_INTGRTR_ERROR);

//Compute cumulative Integrator term IntegPres[FR]
=INTGRAL_PRESSR[FR]+(IntegralError[FR]*K_I*LOOP
_TIME_SEC);

INTGRAL_PRESSR[FR]=IntegPres[FR];

//Limit integral term to be as negative as the P,
D, DD sum is positive, so as not to delay //a fast
```

ramping positive P+D+DD pressure

```
if ((INTGRAL_PRESSR[FR]<0)&&(P_D_DD_Sum[FR]>0)
&&(INTGRAL_PRESSR[FR]<(-P_D_DD_Sum[FR]))) )
    INTGRAL_PRESSR[FR]=-P_D_DD_Sum[FR];

//If both integral pressure and PD sum are
    negative, zero out the integral pressure term

//so as not to delay total req. pressure from
    ramping up when the PD sum does become
    positive

else if ((INTGRAL_PRESSR[FR]<0)&&(P_D_DD_Sum
[FR]<0)) INTGRAL_PRESSR[FR]=0;
```

**[00215]** For right turns (negative roll angle), the integral pressure term for outer (left) front wheel is determined by:

```
//Compute raw value of Integral error for right
    wheel IntegralError[FL]=ROLL_SIG_FOR_CONTROL
    +INTG_DB;

//Put bounds on min and max values of the integral
    error, to prevent integrator windup Integral
    Error[FL]=LIMIT(IntegralError[FL],((-1)*MAX
    _INTGRTR_ERROR), MIN_INTGRTR_ERROR);

//Compute cumulative Integrator term IntegPres[FL]
    =INTGRAL_PRESSR[FL]-(IntegralError[FR]
    *K_I*LOOP_TIME_SEC);

INTGRAL_PRESSR[FL]=IntegPres[FL];

// Limit integral term to be as negative as the P,
    D, DD sum is positive, so as not to delay

// a fast ramping positive P+D+DD pressure

if ((INTGRAL_PRESSR[FL]<0)&&(P_D_DD_Sum
[FL]>0)&&(INTGRAL_PRESSR[FL]<(-P_D_DD_Sum
[FL]))) INTGRAL_PRESSR[FL]=-P_D_DD
_Sum[FL];

//If both integral pressure and PD sum are
    negative, zero out the integral pressure term //so
```

as not to delay total req. pressure from ramping up when the PD sum does become positive

```
else if ((INTGRAL_PRESSR[FL]<0)&&(P_D_DD_Sum[FL]
<0))INTGRAL_PRESSR[FL]=0;
```

**[00216]** In step 416 the ramp down rate of PID requested pressure while PID intervention is active is limited. The hydraulic control unit can release pressure faster than it can build it. So to provide for a more uniform PID pressure request, the ramp down rate of PID pressure is limited to the symmetrical value of the hydraulics build rate. This helps keep a consistent level of slip on the control wheel which in turn reduces possible roll oscillations. This is performed by the following logic:

```
for (WheelIndex=FL; WheelIndex<=FR; WheelIndex++)
{
//Calculate upstream unlimited delta pressure
request Raw_DeltaP[WheelIndex]=INTGRAL_PRESSR
[WheelIndex]+P_D_DD_Sum[WheelIndex]-Z1_PID_DD_SUM
[WheelIndex];
//Conditions for entering and keeping in ramp down
limit logic if (PID_ACTIVE[WheelIndex]&&(((Raw
_DeltaP[WheelIndex]<MAX_PRESSURE_DECREASE_TO
_ENTER_LMTD_RAMP_DOWN)&&!LIMITED_RAMP_DOWN
_RATE_MODE[WheelIndex])
||((Raw_DeltaP[WheelIndex]<MIN_PRESSURE_INCREASE_TO
_STAY_IN_LMTD_RAMP_DOWN)&&LIMITED_RAMP_DOWN
_RATE_MODE[WheelIndex])))
{
//Set limited ramp down rate mode to true LIMITED
_RAMP_DOWN_RATE_MODE[WheelIndex] =TRUE;
LmtdRampDownRate=LMTD_RAMP_DOWN_RATE;

//If opposite front wheel starts requesting PID
pressure, override ramp down rate by mirror value
of max
//hydraulics build rate if (PID_ACTIVE[CROSS_AXLE
_WHEEL(WheelIndex)])
{
```

```

LmtdRampDownRate=MAX_RAMP_DOWN_RATE;
}
//calculate delta pressure decrease term for
current loop, by capping lower limit to
LmtdRampDownRate
//Given that the delta will be negative, take the
maximum of it and the ramp down rate limitLmtd
_DeltaP_For_RmpDwn[WheelIndex]=max(Raw_DeltaP
[WheelIndex], LmtdRampDownRate);
}

else //Limited ramp down rate mode was not entered
{
    //Set limited ramp down rate mode to false
state
    LIMITED_RAMP_DOWN_RATE_MODE[WheelIndex]=FALSE;
}
}

```

**[00217]** In step 418 the total PID requested pressure of the current loop is determined according to the following four modes:

```

for (WheelIndex=FL; WheelIndex<=FR; WheelIndex++)
{
    //Save P,I,D,DD sum for next loopZ1_PID_DD
    _SUM[WheelIndex]=P_D_DD_Sum[WheelIndex]
    +INTGRAL_PRESSR[WheelIndex];

    (1) If in ramp down mode, decrement by limited ramp
down delta if (LIMITED_RAMP_DOWN_RATE_MODE
[WheelIndex])
    {
        PIDStblzPres[WheelIndex]=PID_STBLZ_PRES[Wheel
Index]+Lmtd_DeltaP_For_RmpDwn[WheelIndex];
    }

    (2) If total increasing PID pressure is positive
and PID control is active, limit pressure increase
delta to take into account actuator drive limits.
The limit becomes effective once pressure estimate
increases by PRESSURE_INCREASE_DELTA_FOR_INFLECTION
_ADJUST from initial pressure estimate value upon
activation, then will remain in effect as long as
pressure estimate remains above the exit threshold
(EXIT_THRES)

```

```

else if (PID_ACTIVE [WheelIndex]
&&((P_D_DD_Sum[WheelIndex]+INTGRAL_PRESSR
[WheelIndex])>0)
&&(((BRAKE_PRESSR_ESTMT[WheelIndex]-PRES
EST_UPON_PID_INCREASE[WheelIndex])>
PRESSURE_INCREASE_DELTA_FOR_INFLECTION_ADJUST)
&&(!INITIAL_ADJUST_HAS_OCCURRED[WheelIndex]))
||((BRAKE_PRESSR_ESTMT[WheelIndex]>EXIT_THRES)
&&(INITIAL_ADJUST_HAS_OCCURRED[WheelIndex])))
)
{
if ((PID_STBLZ_PRES[WheelIndex]+Raw_DeltaP[Wheel
Index]-BRAKE_PRESSR_ESTMT[WheelIndex])
>PRESSURE_OFFSET_DURING_PID_RAMP_UP)
{
PIDStblzPres[WheelIndex]=BRAKE_PRESSR_ESTMT
[WheelIndex]+ PRESSURE_OFFSET_DURING_PID
_RAMP_UP;
}
else
{
PIDStblzPres[WheelIndex]=PID_STBLZ_PRES[Wheel
Index]+Raw_DeltaP[WheelIndex];
}

//Logic for detecting initial adjustment of
req. pressure towards pressure estimate during
ramp up phase
if (!INITIAL_ADJUST_TO_PRES_ESTMT [WheelIndex]
&&!INITIAL_ADJUST_HAS_OCCURRED[WheelIndex])
{
INITIAL_ADJUST_TO_PRES_ESTMT [WheelIndex]
=TRUE;
INITIAL_ADJUST_HAS_OCCURRED[WheelIndex]
=TRUE;
}
//Set INITIAL_ADJUST_TO_PRES_ESTMT flag for
only one loop's duration per activation cycle
else
INITIAL_ADJUST_TO_PRES_ESTMT [Wheel
Index]=FALSE;
}

```

(3) If total increasing PID pressure is positive and PID control is active, but pressure estimate has not yet increased over the activation pressure estimate by PRESSURE\_INCREASE\_DELTA\_FOR\_INFLECTION



\_ADJUST, assign total PID pressure to prior total plus the upstream unlimited delta pressure increase.

```

else if (PID_ACTIVE [WheelIndex]&&((P_D_DD_Sum
[WheelIndex]+INTGRAL_PRESSR[WheelIndex])>0))
{
    PIDStblzPres[WheelIndex]=PID_STBLZ_PRES[Wheel
Index]+Raw_DeltaP[WheelIndex];
}
(4)If PID control is active but underlying P,I,D,DD
pressure sum is negative, OR PID is not active, the
total PID pressure is assigned to the underlying
P,I,D,DD pressure sum.

else
    PIDStblzPres[WheelIndex]=P_D_DD_Sum[Wheel
Index]+INTGRAL_PRESSR[WheelIndex];

    PIDStblzPres[WheelIndex]=LIMIT(PIDStblzPres
[WheelIndex], 0, MAXIMUM_STBLZ_PRESSURE);

} /*End for loop*/

```

**[00218]** The PID control entrance and exit strategy is performed in step 420. A history of vehicle lateral acceleration exceeding a given large threshold is kept to be used as a screening criteria for PID activation is performed.

```

if (ABS(CG_FLT_LAT_ACC)>LAT_ACC_ACTVTION_THRSHLD)
    LARGE_LAT_ACC_COUNTER=LAT_ACC_COUNTER_INIT;

else if (LARGE_LAT_ACC_COUNTER>0)
    LARGE_LAT_ACC_COUNTER--;

```

**[00219]** PID control is enabled in the following.

```

if ((LARGE_LAT_ACC_COUNTER>0)
// Large lat acc or recent history of large
lat acc
&&((REF_VELOCITY_YC>PID_MINIMUM_ACTIVATION
_SPEED)||
((PID_ACTIVE [FL]||PID_ACTIVE[FR])&&
(REF_VELOCITY_YC>MAXIMUM_SPEED_TO

```

```

        _CONTINUE_PID))
    )
    &&(!RSC_DISABLED)    //RSC system is enabled,
    no shutdowns exist
    &&(!REVERSE_MOVEMENT)    //Allow activation
    only going forward
    &&(!STATUS_FIRST_RUN)    //Allow activation
    only after first run becomes false
)
EnablePIDControl=TRUE;
else
    EnablePIDControl=FALSE;

```

**[00220]** Activating/deactivating individual wheel PID control, figuring in Driver Brake Apply is determined in step 422. This is set forth in the following code:

```

for (WheelIndex=FL; WheelIndex<=FR; WheelIndex++)
{
    if (PID_ACTIVE [WheelIndex])
    {
        //PID control already active on wheel,
        check for existence of de-activation
        conditions

        if (!EnablePIDControl||
            //Enforce exit when PID pressure
            goes below 9 bar(PIDStblzPres
            [WheelIndex]<=MIN_PID_PRES_FOR
            _FORCED_CONTROL_EXIT)||
            //Once PID control pressure gets
            within ~10 bar of driver pressure
            AND ((PIDStblzPres[WheelIndex]
            <(DRIVER_REQ_PRESSURE+EXIT_THRES))
            && //PID control pressure is less
            than or equal to wheel pressure
            estimate (in case pressure build
            //is in a slow build mode and thus
            not reflecting driver pressure) AND
            */(PIDStblzPres[WheelIndex]<=(BRAKE
            _PRESSR_ESTMT[WheelIndex]))&&
            //wheel slip ratio is greater than~
            -15% to indicate appropriateness for
            ABShandoff */(SLIP_RATIO[WheelIndex]
            >=WHEEL_STABLE_IN_SLIP)))
        {
            PID_ACTIVE[WheelIndex]=FALSE;

```

```

        INTGRAL_PRESSR[WheelIndex]=0;          //Reset key
state variables
        PRES_EST_UPON_PID_INCREASE[WheelIndex]=0;
        REFERENCE_PROP_PEAK_HOLD[WheelIndex]=0;
        PROP_PEAK_HOLD_TIMER[WheelIndex]=0;
        RAMP_DOWN_PROP_HOLD_TO_PROP_PRES[WheelIndex]
        =FALSE;

[00221]   If driver is braking upon PID exit, a flag is
set to desensitize PID re-entry
        if (DRIVER_BRAKING_FLAG)

                EXIT_PID_DURING_DRIVER_BRAK[WheelIndex]=TRUE;
        else

                EXIT_PID_DURING_DRIVER_BRAK[WheelIndex]=FALSE;
        }
        //End PID_ACTIVE==TRUE

    else          //PID control not active on wheel, check
if conditions exist for activation
        {
                //reset initial PID pressure adjust (toward
                pressure estimate) flag for next PID
                activation
                INITIAL_ADJUST_HAS_OCCURRED[WheelIndex]=FALSE;
                //Keep forcing prop peak hold exit until PID
                reactivation by resetting key state variables
                PROP_PEAK_HOLD_TIMER[WheelIndex]=0;
                REFERENCE_PROP_PEAK_HOLD[WheelIndex]=0;
                RAMP_DOWN_PROP_HOLD_TO_PROP_PRES[WheelIndex]
                =FALSE;
                //If driver is no longer braking, allow PID
                entry at entrance threshold
                if (!DRIVER_BRAKING_FLAG)
                {
                        EXIT_PID_DURING_DRIVER_BRAK[WheelIndex]
                        =FALSE;
                }
                if (((//If driver is no longer braking OR
                driver is braking but PID intervention had not
                //been called for yet, allow PID entry at
                entrance threshold
                ((PIDStblzPres[WheelIndex]>=ENTER_THRES)&&
                (!EXIT_PID_DURING_DRIVER_BRAK[WheelIndex]))
                //If PID recently exit due to driver braking
                and driver is still braking, allow PID

```

```

intervention once
//PID pressure exceeds pressure estimate plus
a threshold, to provide for hysteresis
||(EXIT_PID_DURING_DRIVER_BRAK[WheelIndex]&&
(PIDStblzPres[WheelIndex]>=ENTER_THRES)
&&(PIDStblzPres[WheelIndex]>=(BRAKE
PRESSR_ESTMT[WheelIndex]+EXIT_THRES))
//Can also enter at lower threshold if had
recent PID event and driver is not braking
|(((PIDStblzPres[WheelIndex]>=EXIT_THRES)&&REC
ENT_PID_CTRL&&(!DRIVER BRAKING_FLAG))
)&&EnablePIDControl
)
{
PID_ACTIVE[WheelIndex]=TRUE;
//Latch pressure estimate at PID activation,
for use in PID pressure adjustment toward
estimate during ramp up phase
PRES_EST_UPON_PID_INCREASE[WheelIndex]
=BRAKE_PRESSR_ESTMT[WheelIndex];
//If driver is braking, include the minimum of
driver pressure or pressure estimate into the
PID pressure state
PIDStblzPres[WheelIndex]+=min(DRIVER_REQ
PRESSURE, BRAKE_PRESSR_ESTMT[WheelIndex]);

//Put an upper and lower bound on the final
PID requested pressure value
PIDStblzPres[WheelIndex]=LIMIT(PIDStblz
Pres[WheelIndex], 0, MAXIMUM_STBLZ_PRESSURE);
}
}

```

**[00222]** The PID pressure is then applied to the brake system to prevent the vehicle from rolling over. The change in the current loop's delta requested pressure is also determined.

```

DeltaPIDStblzPres[WheelIndex]=PIDStblzPres
[WheelIndex]-PID_STBLZ_PRES[WheelIndex];

//Update INCREASE_REQUESTED_PRESSURE flag
based on previous loop's delta pressure value

if ((DELTA_PID_STBLZ_PRES[WheelIndex]>=0)

```

```

        //Do not set flag when req pressure
        is less than enter threshold value
        &&(PIDStblzPres[WheelIndex]>ENTER
        _THRES)
    )
        INCREASE_REQUESTED_PRESSURE[Wheel
        Index]=TRUE;
    else
        INCREASE_REQUESTED_PRESSURE[Wheel
        Index]=FALSE;

```

**[00223]** Global variables with local versions are updated in step 426.

```

    PID_STBLZ_PRES[WheelIndex]=PIDStblzPres[Wheel
    Index];

    DELTA_PID_STBLZ_PRES[WheelIndex]=DeltaPIDStblzPres
    [WheelIndex];
}    /*End for loop*/

```

**[00224]** PID requested pressure if corresponding flag is false is set to zero in the following:

```

    if (!PID_ACTIVE[FL])PID_STBLZ_PRES[FL]=0;

    if (!PID_ACTIVE[FR])PID_STBLZ_PRES[FR]=0;

```

**[00225]** A history of when the last PID control is kept in the following:

```

    if (PID_ACTIVE [FL]||PID_ACTIVE[FR])RECENT_PID_EVNT
    _CNTR=RECENT_PID_EVNT_THRSHLD;

    else if (RECENT_PID_EVNT_CNTR>0)RECENT_PID_EVNT
    _CNTR--;

    if (RECENT_PID_EVNT_CNTR>0)RECENT_PID_CNTRL_EVENT
    =TRUE;
    else
        RECENT_PID_CNTRL_EVENT=FALSE;

```

**[00226]** A history of when last in Active Yaw Control is also kept in the following:

```

if
  (AYC_IN_CYCLE) RECENT_AYC_EVNT_CNTR=RECENT_AYC_EVNT
    _THRSHLD;

else if (RECENT_AYC_EVNT_CNTR>0) RECENT_AYC_EVNT
  _CNTR--;

if (RECENT_AYC_EVNT_CNTR>0) RECENT_AYC_CNTRL_EVENT
  =TRUE;

else
  RECENT_AYC_CNTRL_EVENT=FALSE;

```

**[00227]** While the invention has been described in connection with one or more embodiments, it should be understood that the invention is not limited to those embodiments. On the contrary, the invention is intended to cover all alternatives, modifications, and equivalents, as may be included within the spirit and scope of the appended claims.